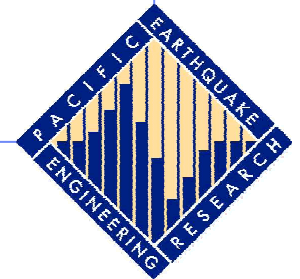


Uniaxial Material Model for Reinforcing Steel Incorporating Buckling and Low-Cycle Fatigue

Jon Mohle and Sashi Kunnath

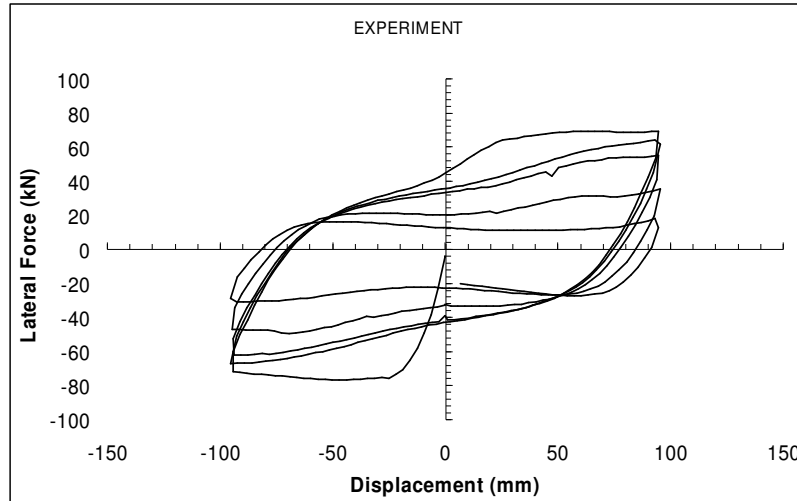


Background and Motivation

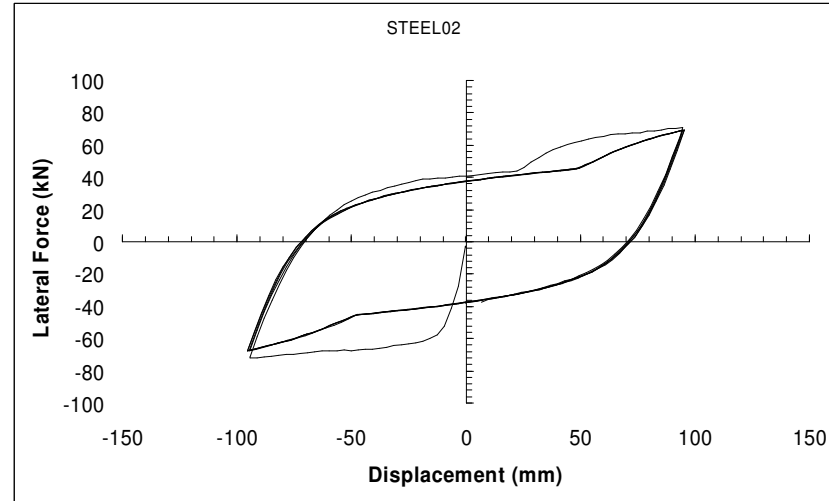
- ◆ Simulation of degrading behavior of RC structures to enable the prediction of post-yield damage states through collapse is critical to PEER performance-based methodology.
- ◆ Use of existing steel material models (Steel01, Steel02) in OpenSees within the framework of a fiber section does not capture degrading behavior

Background and Motivation

Observed response of a bridge column subjected to constant-amplitude cyclic loading

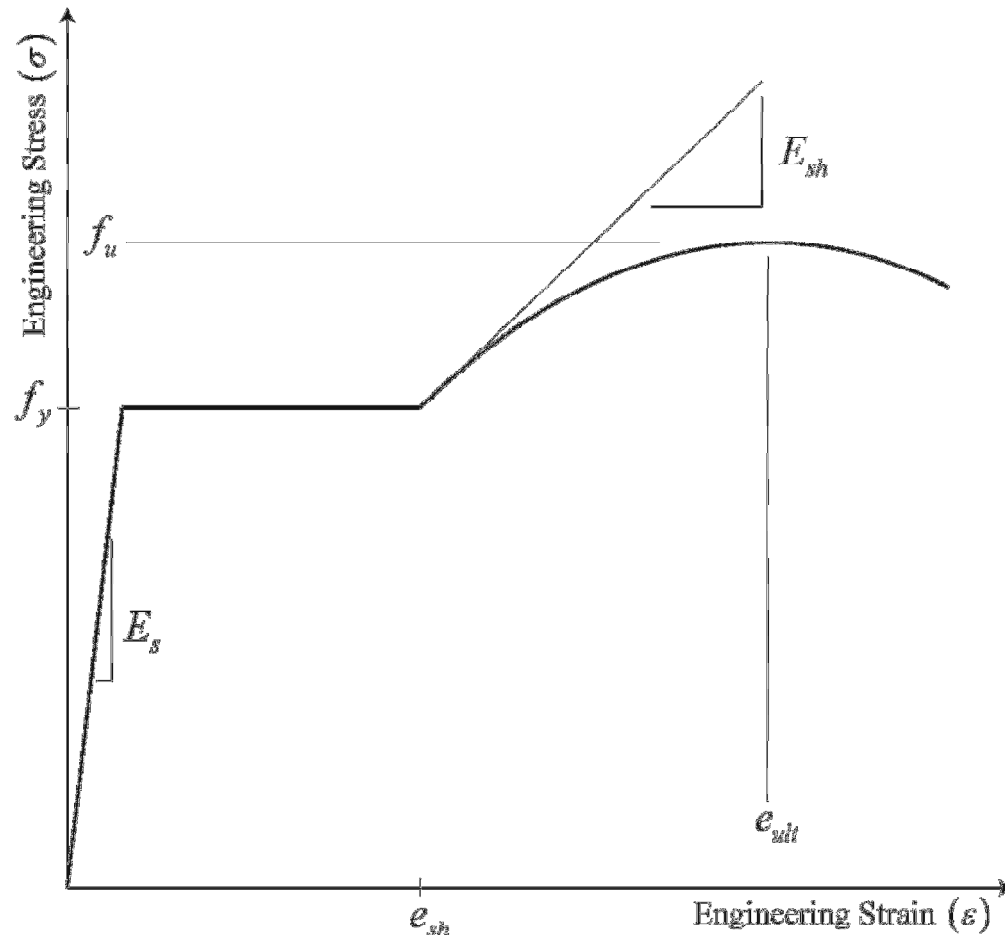


Simulated response using OpenSees with Steel02



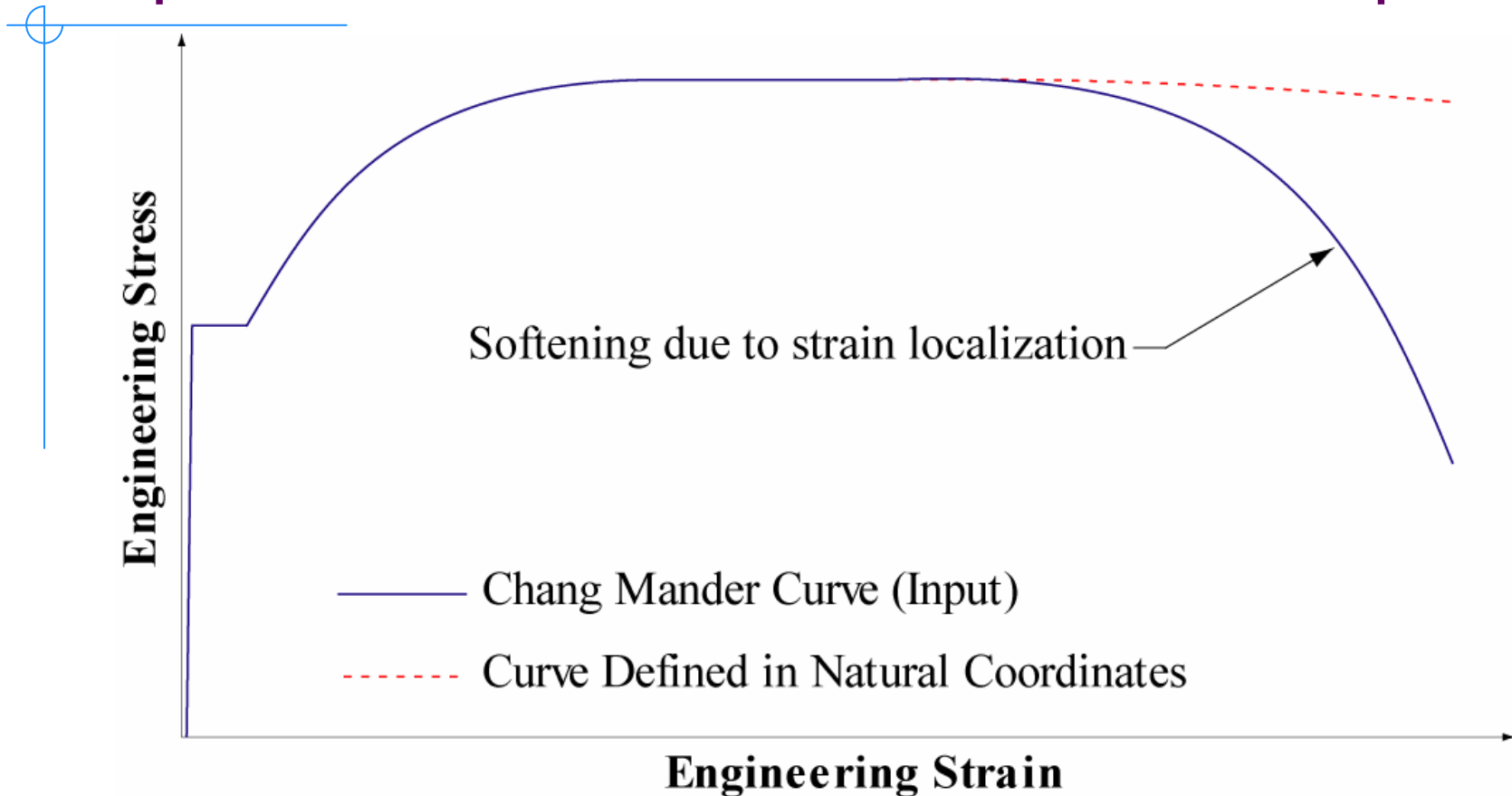
Need for a material model that incorporates buckling, strength deterioration, and failure resulting from low-cycle fatigue

Stress-Strain Envelope



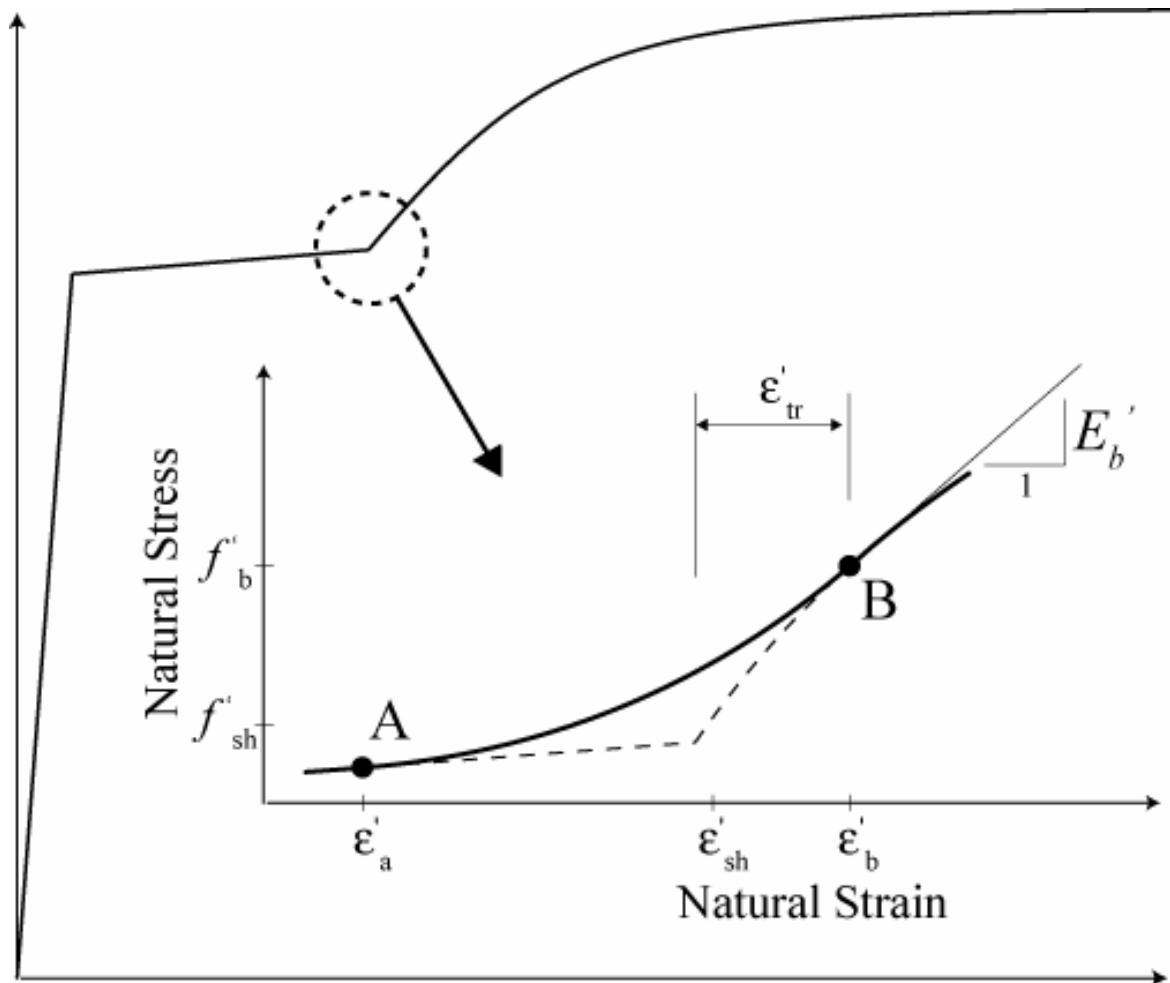
- Note: Envelope is non-symmetric in tension and compression
- Compression envelope is generated automatically – accounting for change in effective bar area
- Cyclic behavior adapted from Chang & Mander (1994)

Representation in Natural Stress-Strain Space



- Tensile Curve Converted to Natural Stress-Strain Space
- Single Curve Represents Both Tension and Compressive Response (tensile and compressive response non-symmetric)

Smoothed Hardening Transition

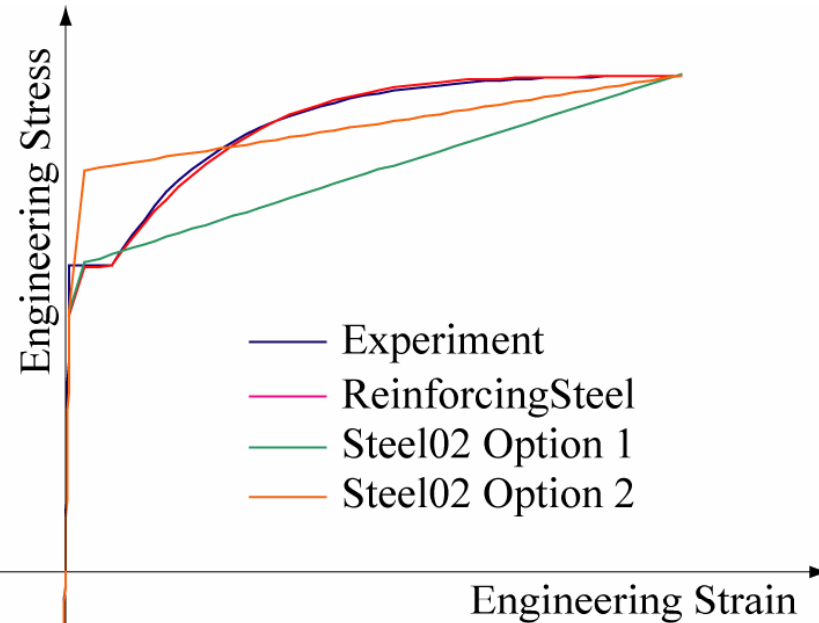


- Continuous slope for improved convergence

Monotonic Material Model Calibration

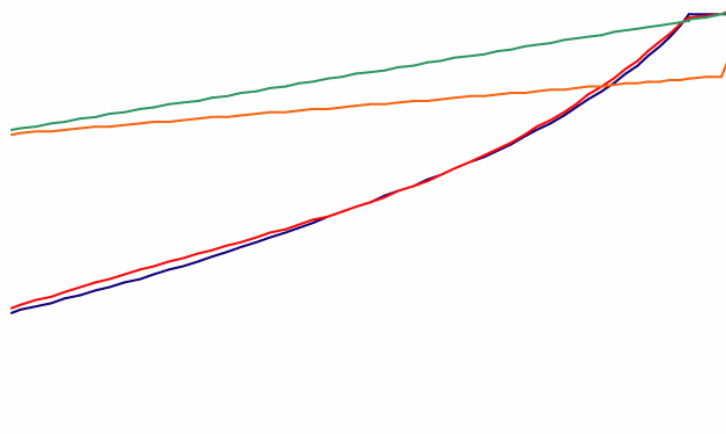
Tensile Response

- Parameters calibrated as closely as possible

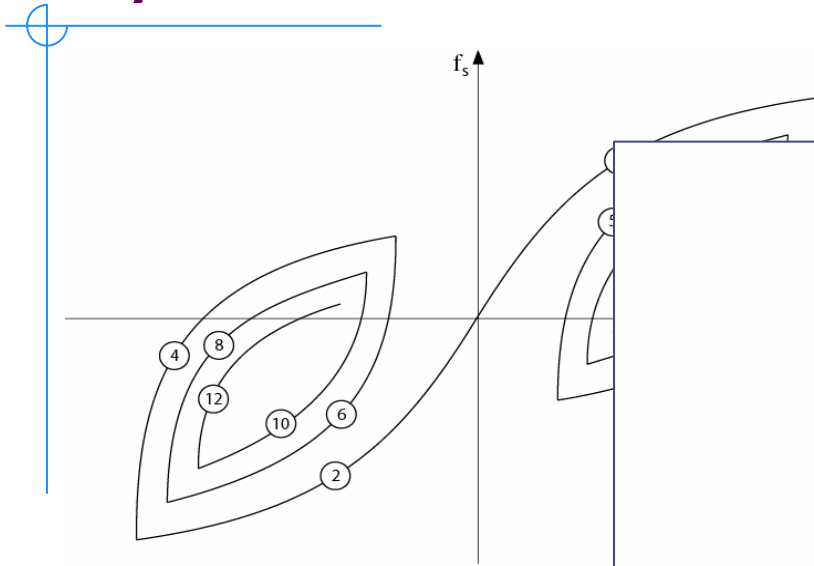


Compressive Response

- Generated using parameters from tensile calibration

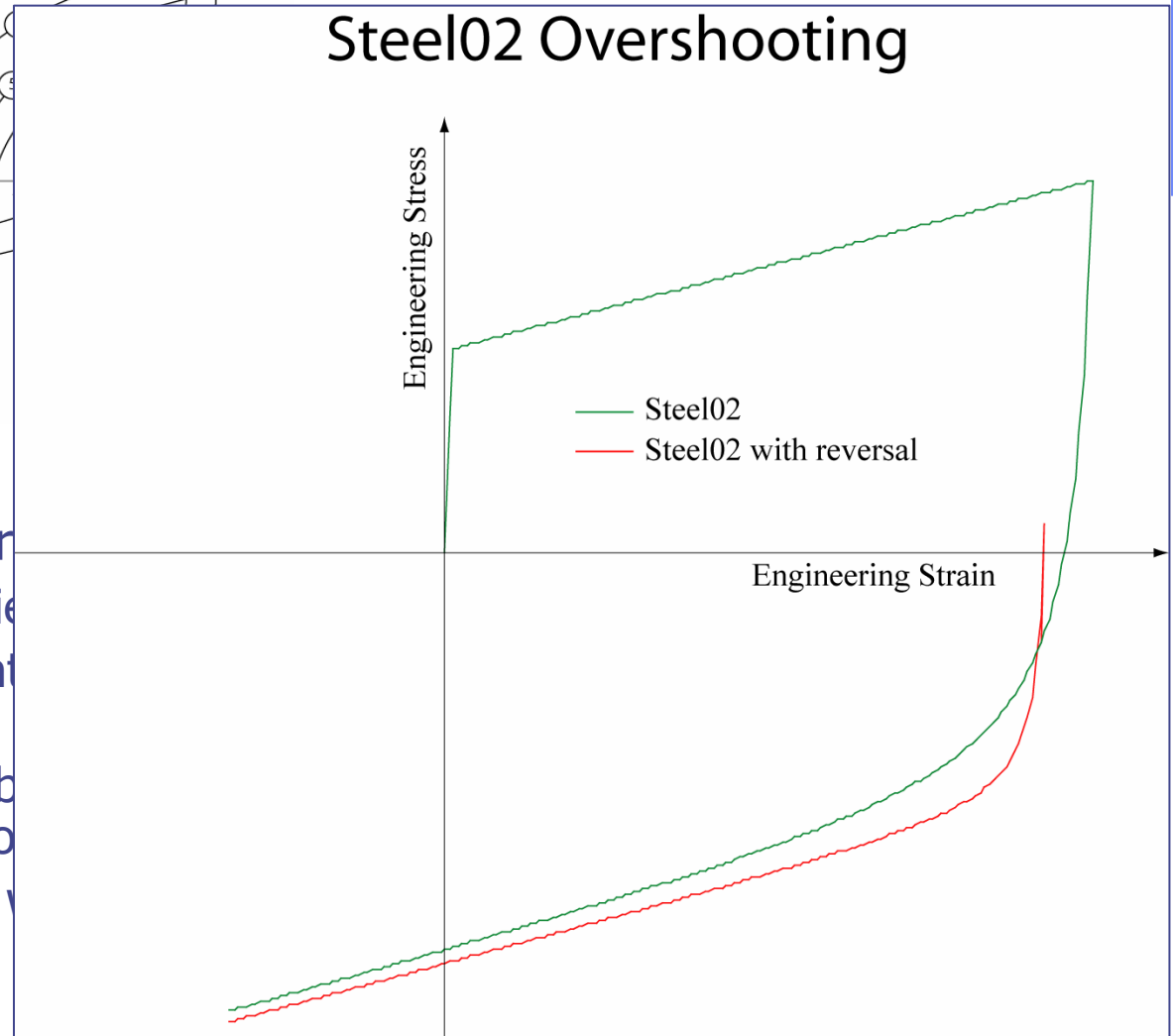


Cyclic Behavior



- Steel02 has the Equivalent of 4 reversal branches

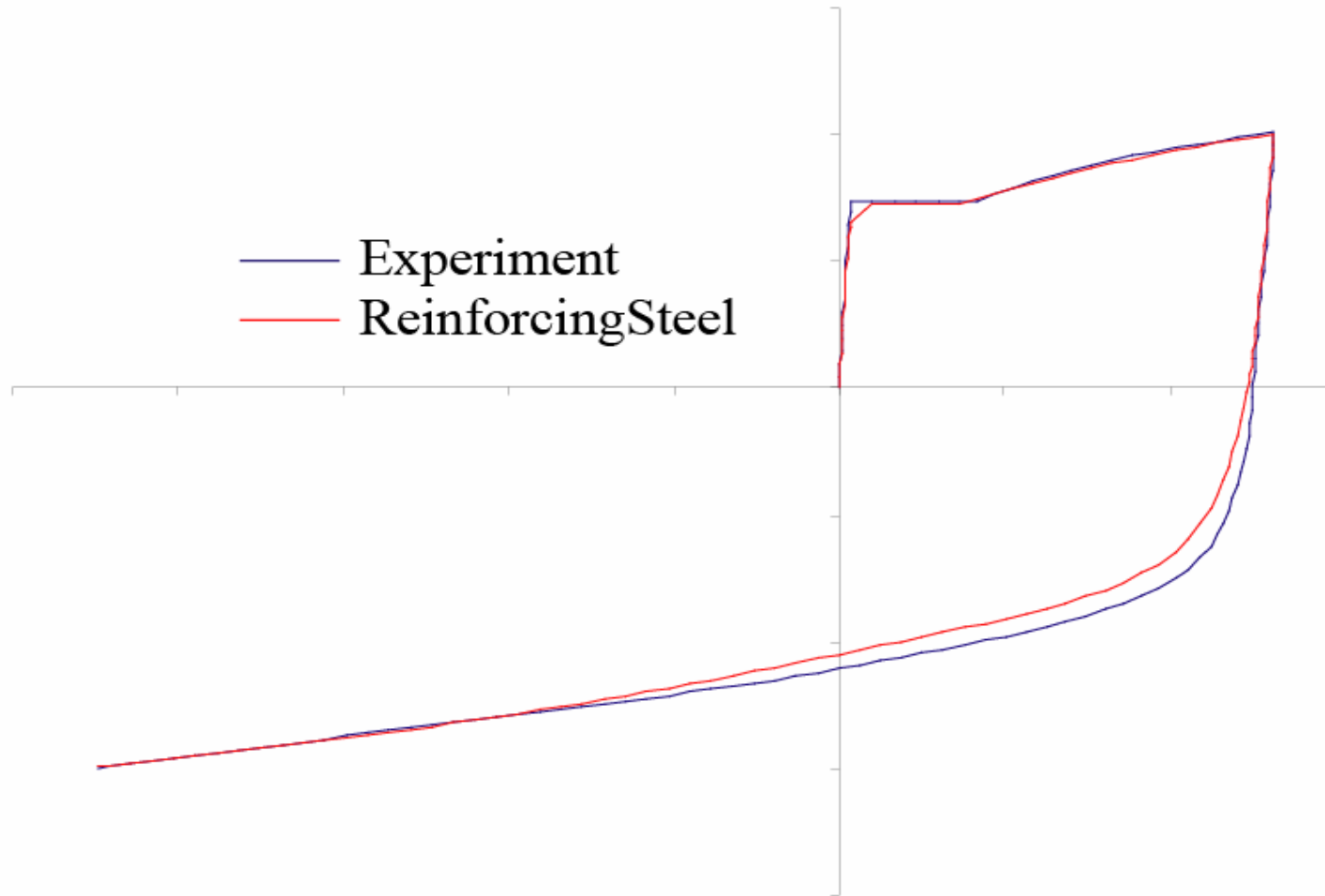
Steel02 Overshooting



- Reasons for added branches
 - Ran EQ time histories
 - Observed significant overshooting
 - Having at least 16 branches minimized overshooting
 - Currently compiled with

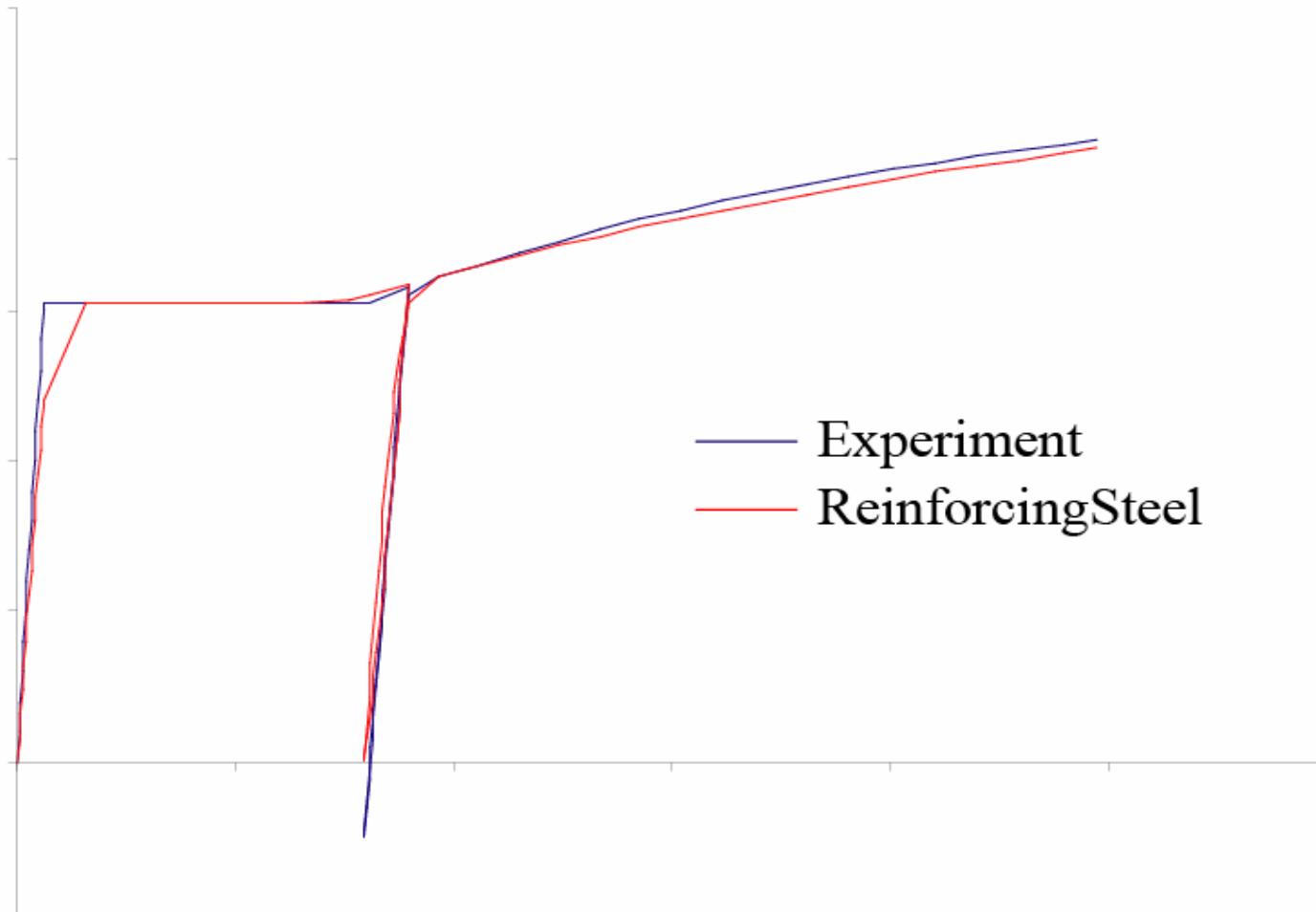
Calibration of Menegotto Pinto Constants

Dodd and Cooke Fig 2.40



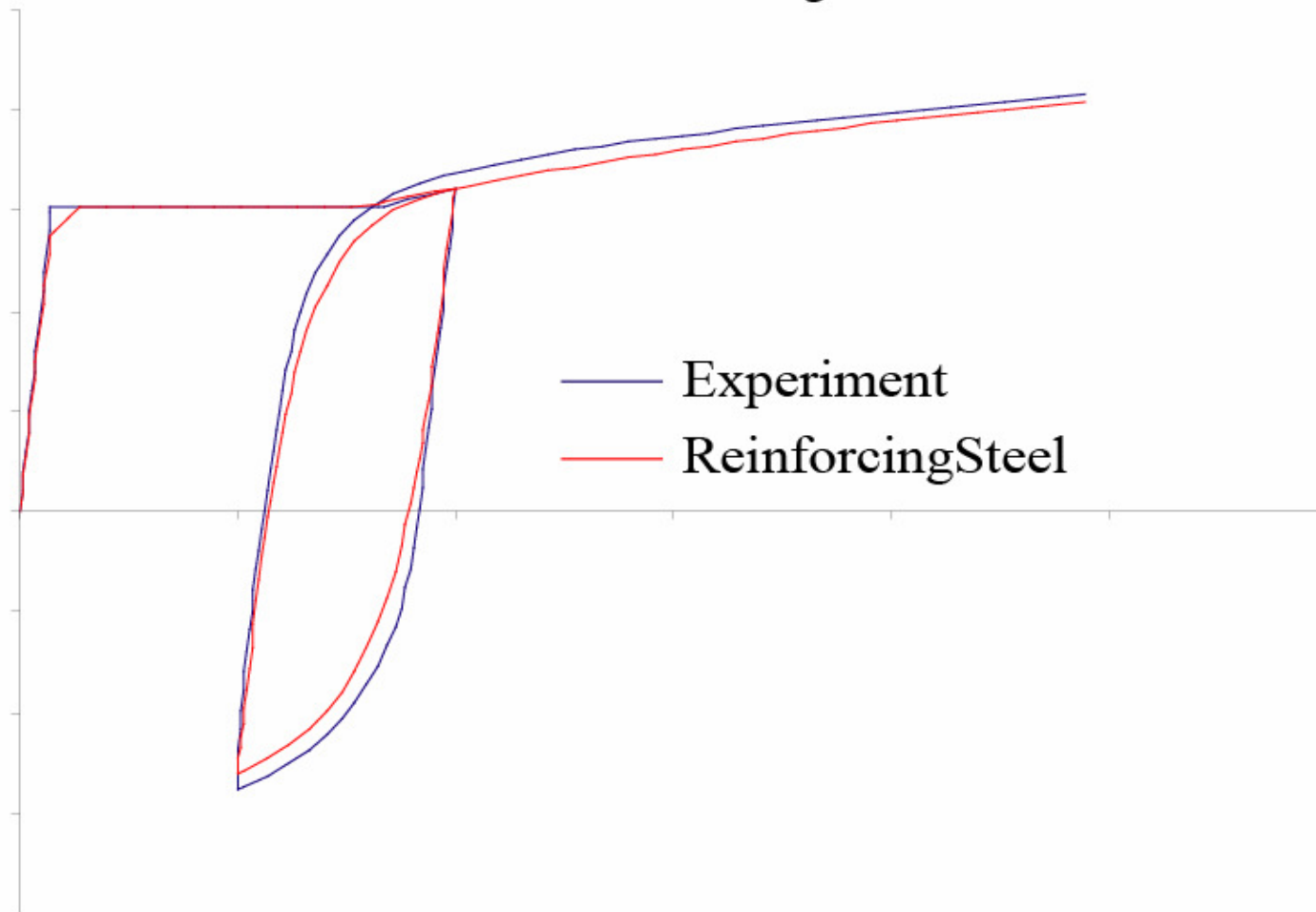
Calibration of Menegotto Pinto Constants

Dodd and Cooke Fig 2.41



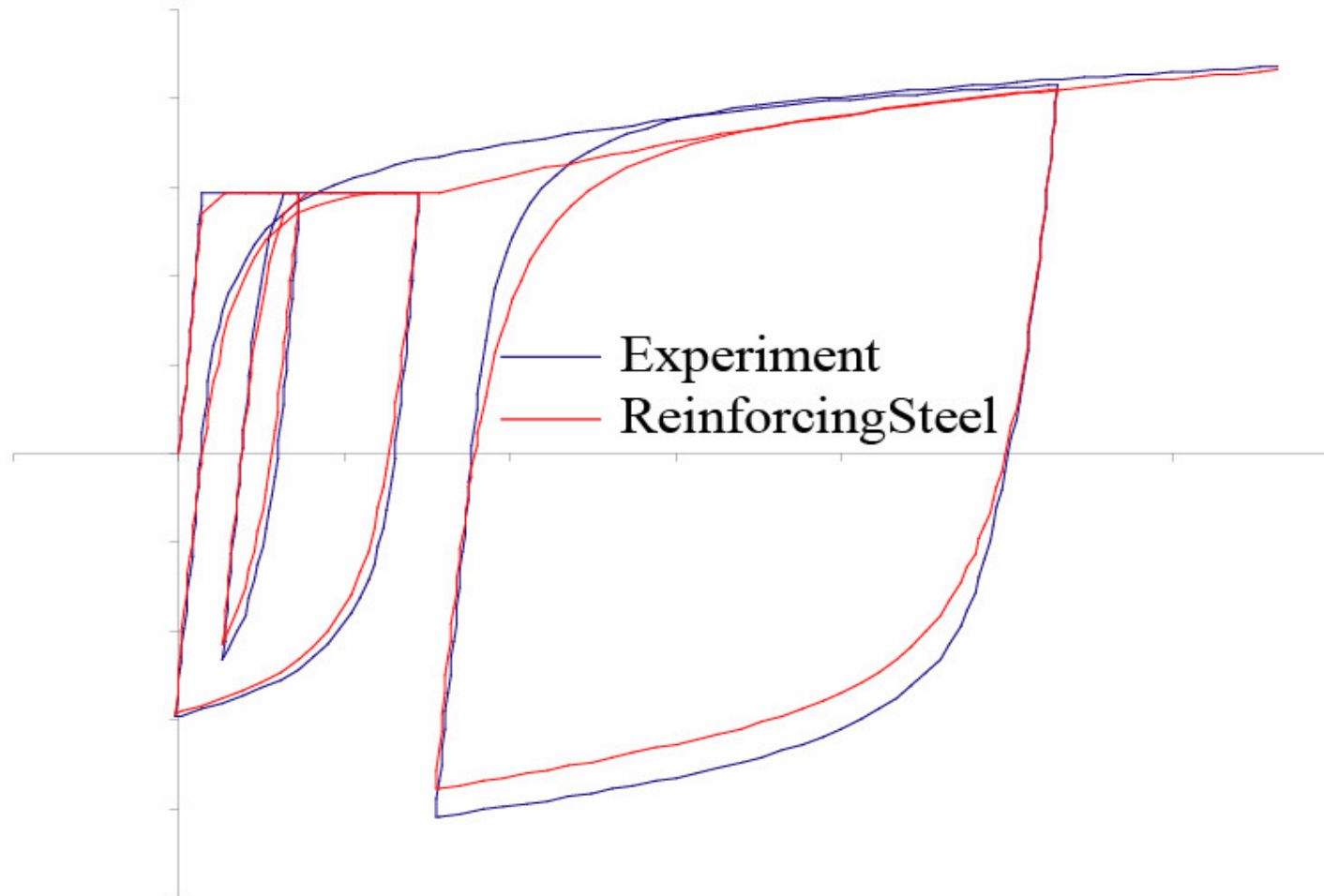
Calibration of Menegotto Pinto Constants

Dodd and Cooke Fig 2.42



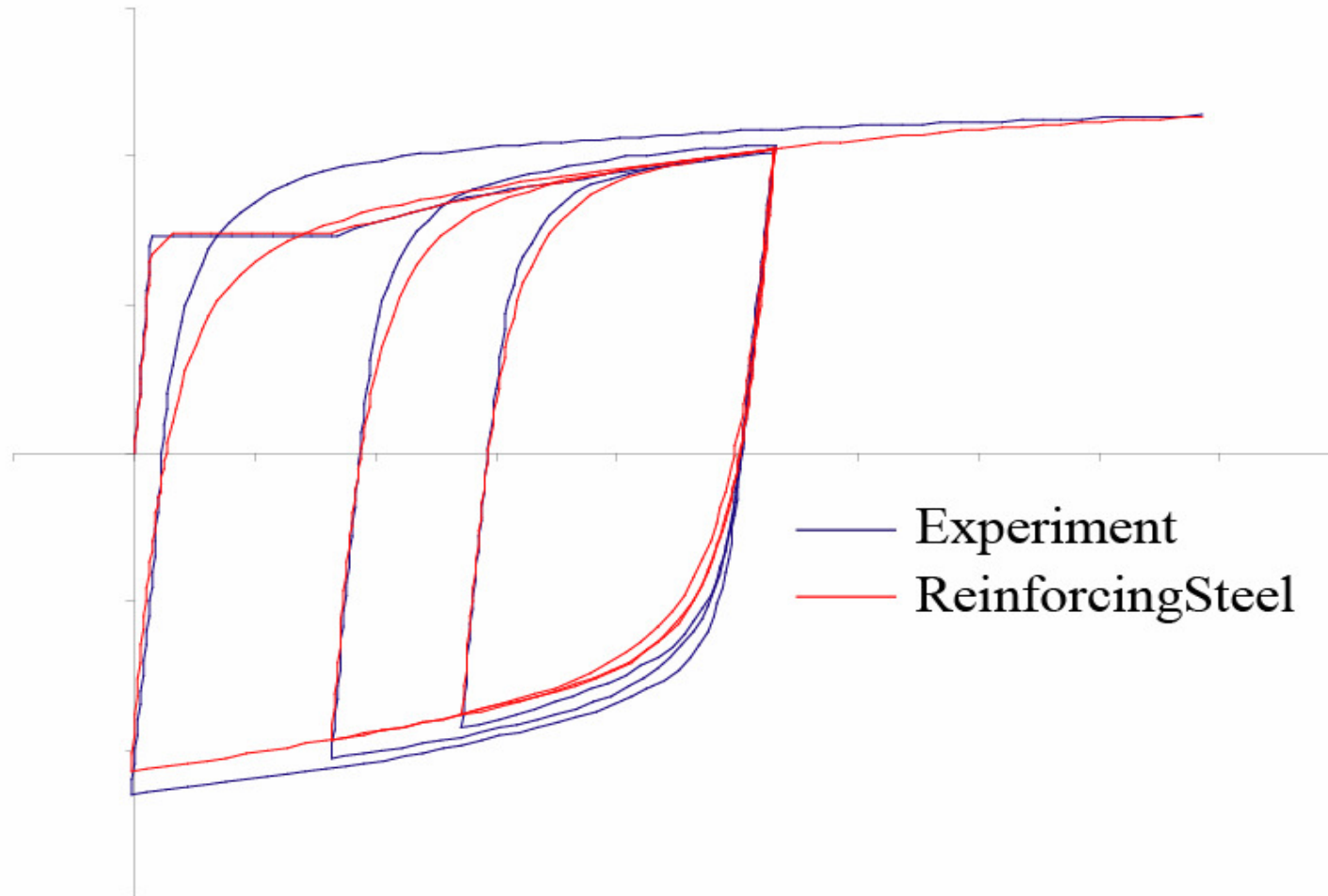
Calibration of Menegotto Pinto Constants

Dodd and Cooke Fig 2.43



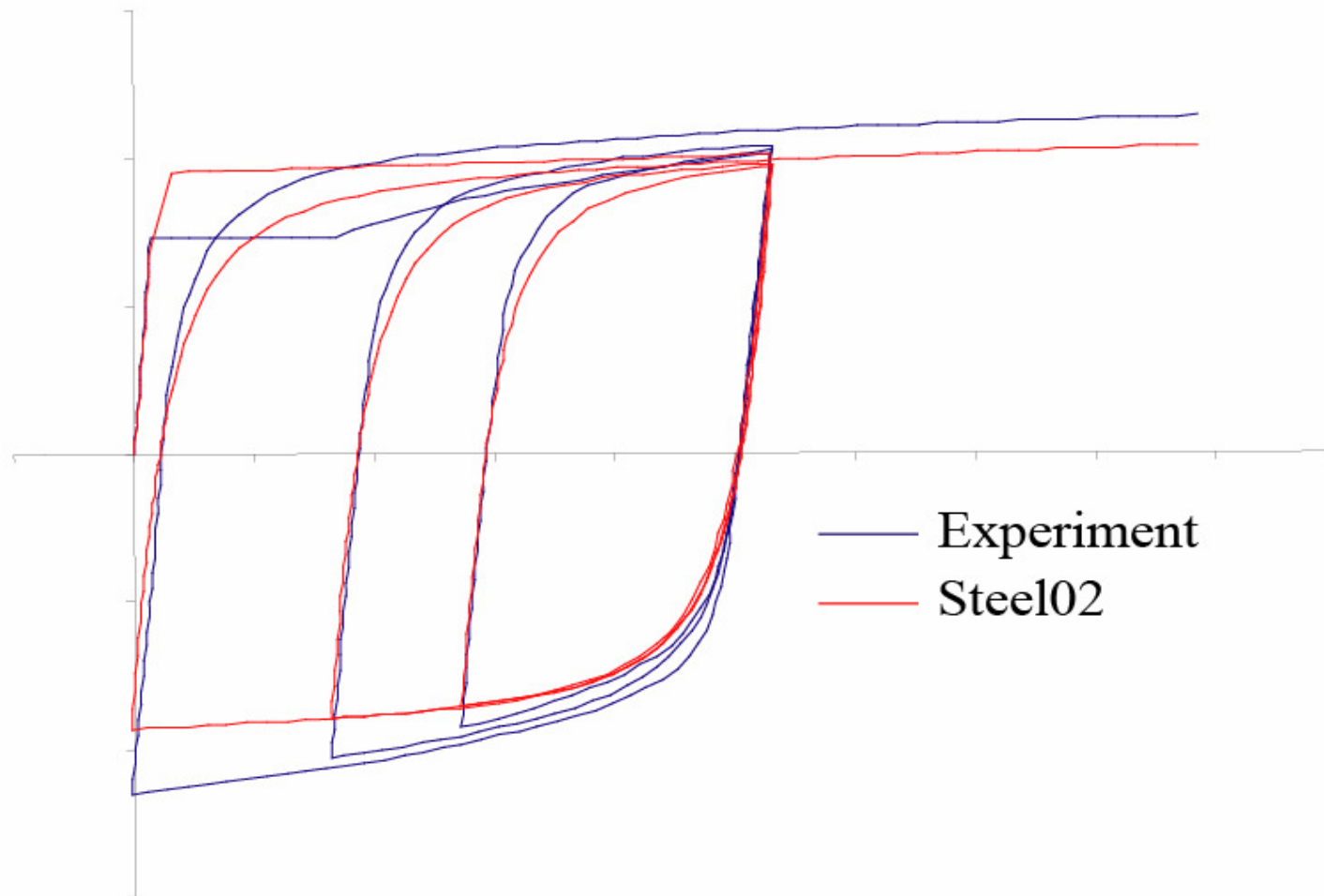
Calibration of Menegotto Pinto Constants

Dodd and Cooke Fig 2.44



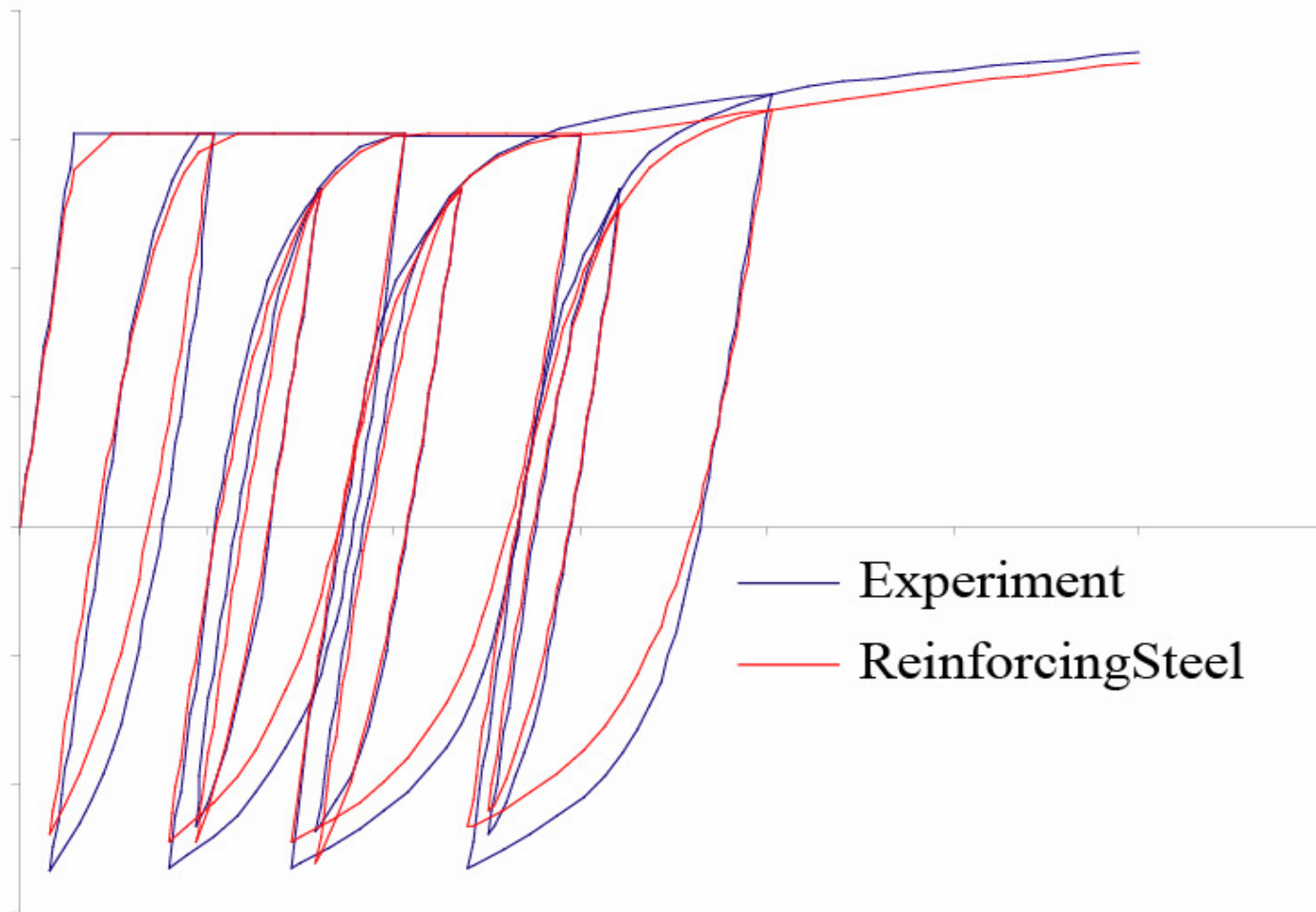
Steel02 (default) No Isotropic Hardening

Dodd and Cooke Fig 2.44



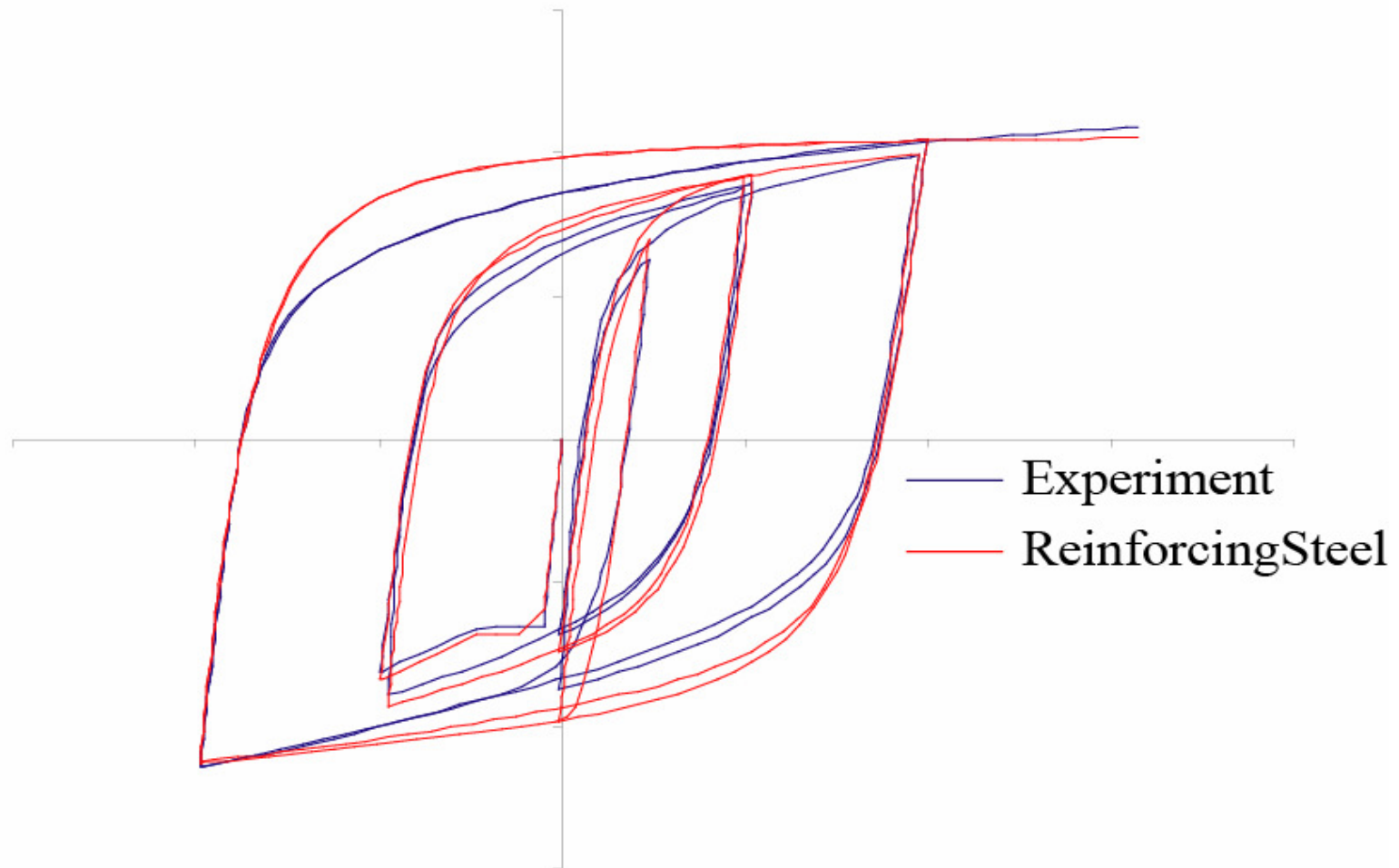
Calibration of Menegotto Pinto Constants

Dodd and Cooke Fig 2.45



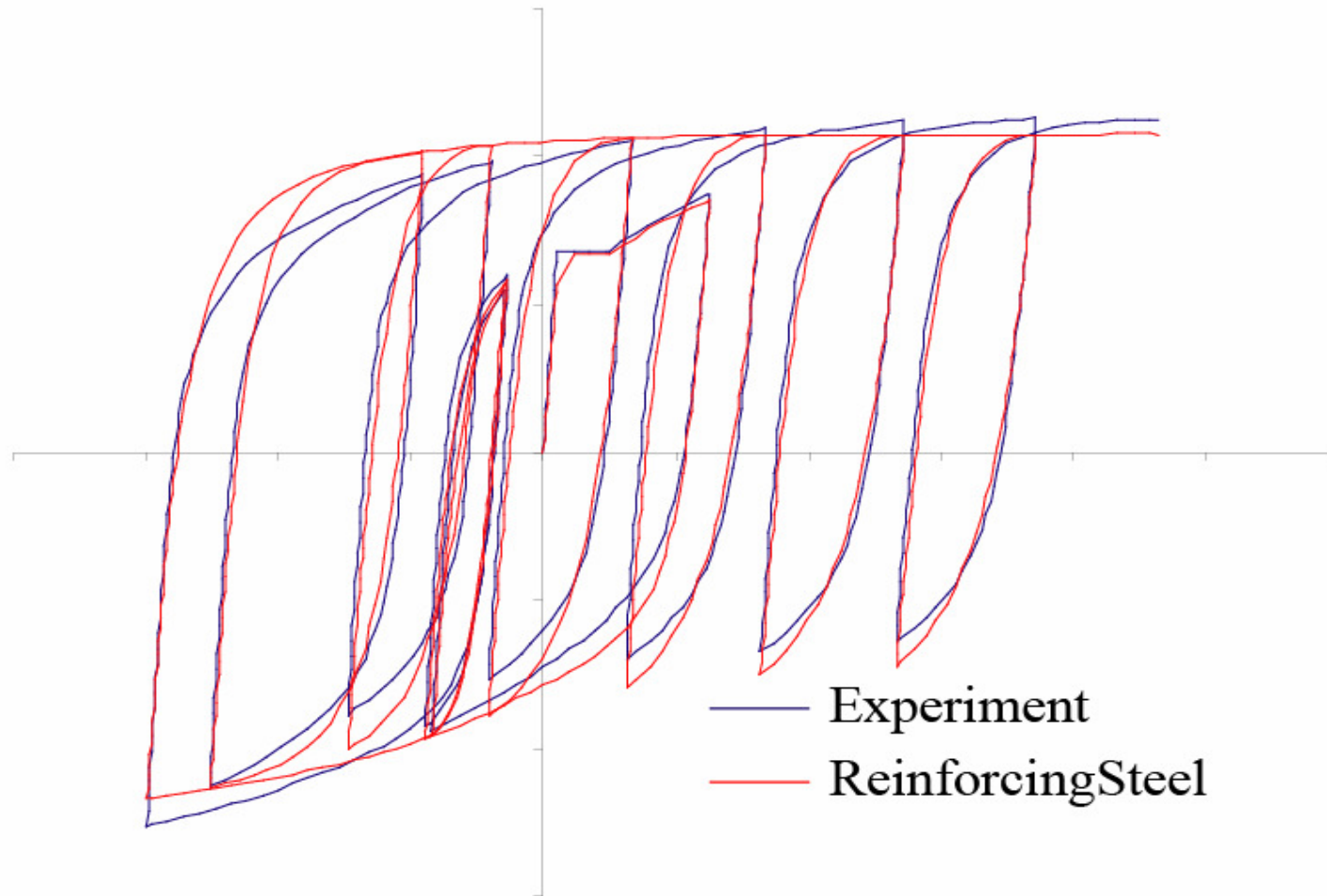
Calibration of Menegotto Pinto Constants

Aktan Test 4

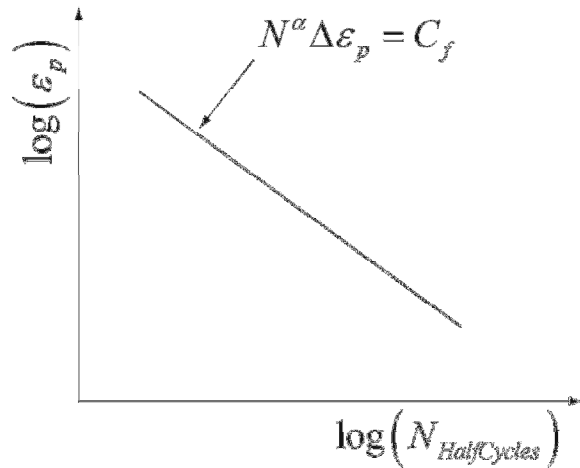


Calibration of Menegotto Pinto Constants

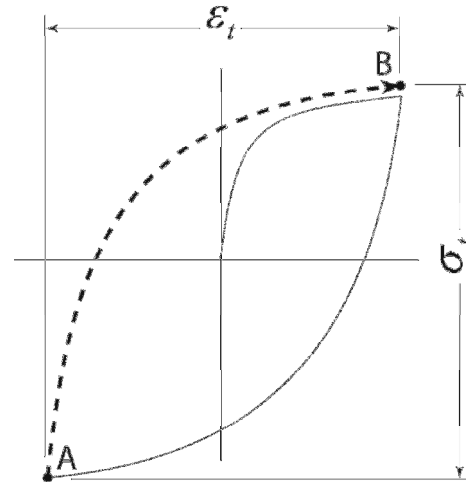
Aktan Test 6



Low-Cycle Fatigue and Fracture: Notation



Coffin-Manson Constants



Half-cycle terms defined

$$\epsilon_p = \epsilon_t - \frac{\sigma_t}{E_s}$$

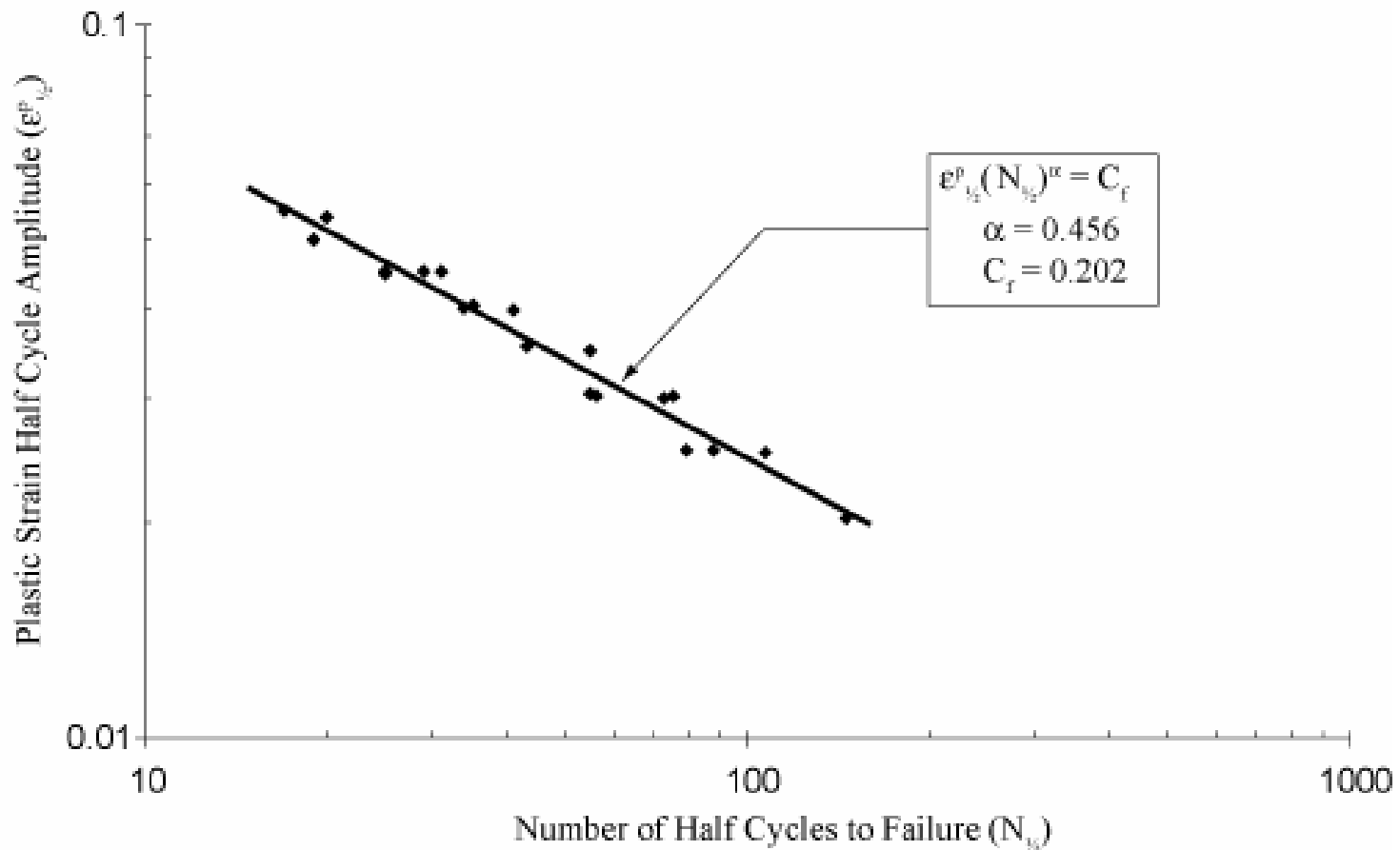
C_f and α are factors used to relate the number of half cycles to fracture to the plastic strain half cycle amplitude. The total half cycle strain amplitude is the change in strain from reversal A to reversal B.

C_f and α are used to define a cumulative damage factor, D:
$$D = \sum \left(\frac{\Delta\epsilon_p}{C_f} \right)^{\frac{1}{\alpha}}$$

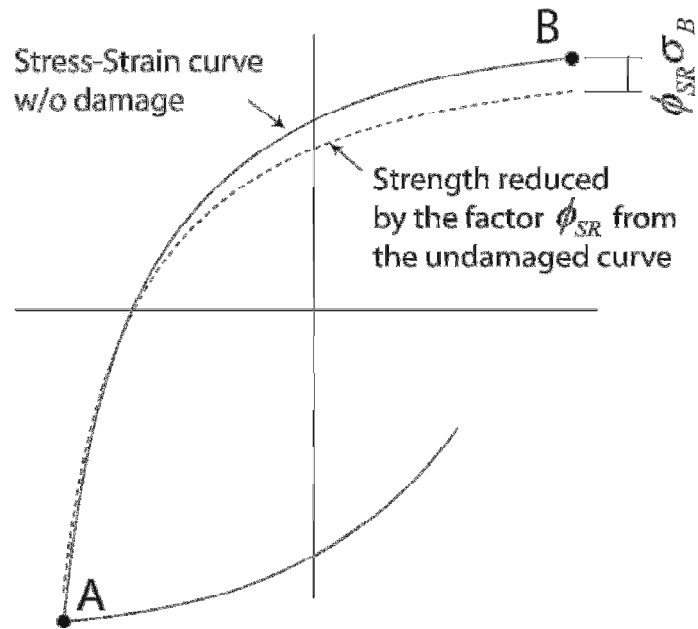
Fatigue Calibration

- Number of cycles to failure vs Average strain amplitude

Coffin-Manson Fatigue Calibration Using Brown and Kunnath Bar Tests



Cyclic Degradation



Another parameter is used to describe loss in strength due to damage or other phenomenon resulting in softening due to plastic reversals. The degradation is currently assumed to have a simple linear relationship with D which correlates strength degradation from an undamaged state to the cumulative damage factor.

$$\phi_{SR} = K_1 D$$

Alternately this simple linear equation can be rewritten in a way that makes the strength degradation independent of the number of half cycles to failure. Keeping the failure and degradation terms independent is convenient for calibration.

$$\phi_{SR} = \sum \left(\frac{\Delta \epsilon_p}{C_d} \right)^{\frac{1}{\alpha}}$$

The constants K_1 and C_d can be related as follows:

$$C_d = \frac{C_f}{K_1^\alpha}$$

Calibration of Cyclic Degradation

$$\phi_{SR} = \sum \left(\frac{\Delta \epsilon_p}{C_d} \right)^{\frac{1}{\alpha}}$$

- Constant α and C_d calibrated using Newton search
- Measured stress was compared with model prediction at peak reversals (constant amplitude tests)
- Minimize Sum of the difference squared
- Calibration results:

- $\alpha=0.451$

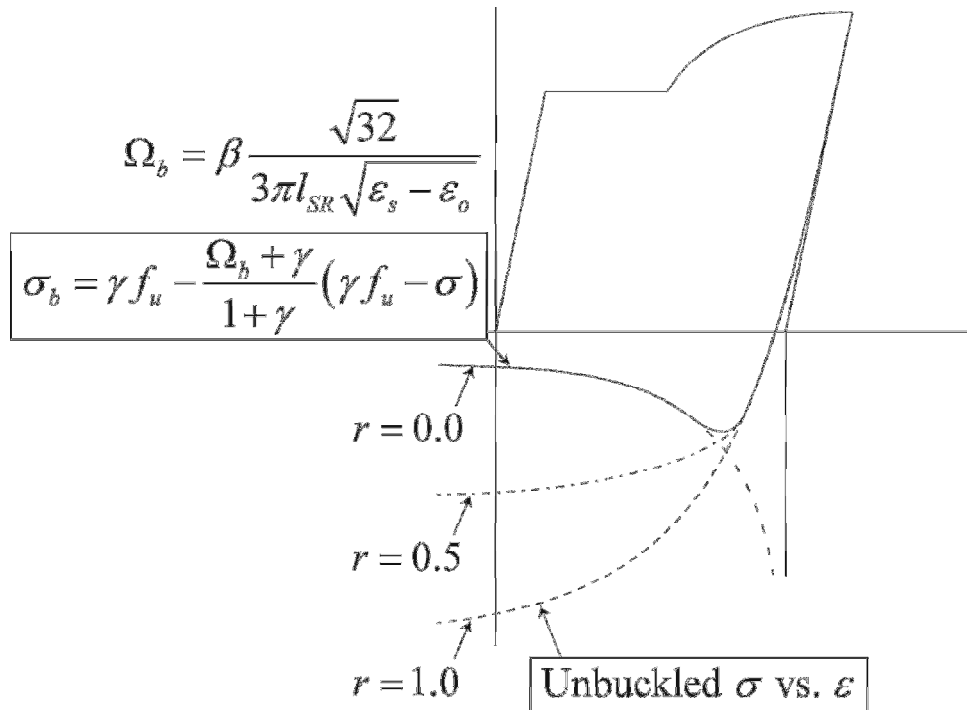
- $C_d=0.477$



$\alpha=0.456$

Note Similarity to Coffin-Manson Calibration

Gomes Appleton Buckling Behavior



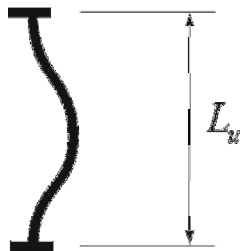
β is an amplification factor to scale the buckling curve.

The r factor adjusts the curve between the buckled and the un buckled profile. The variable r can only be a real number between 0.0 and 1.0.

The γ factor is the positive stress location that the buckling factor is taken about. γ should be between 0.0 and 1.0.

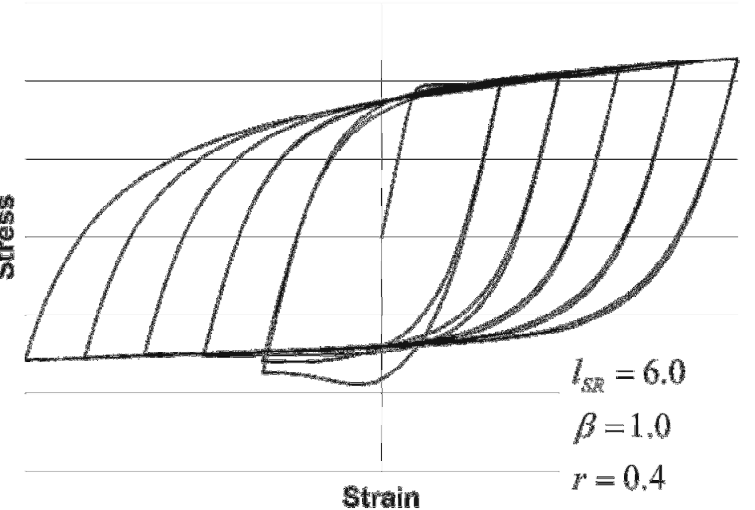
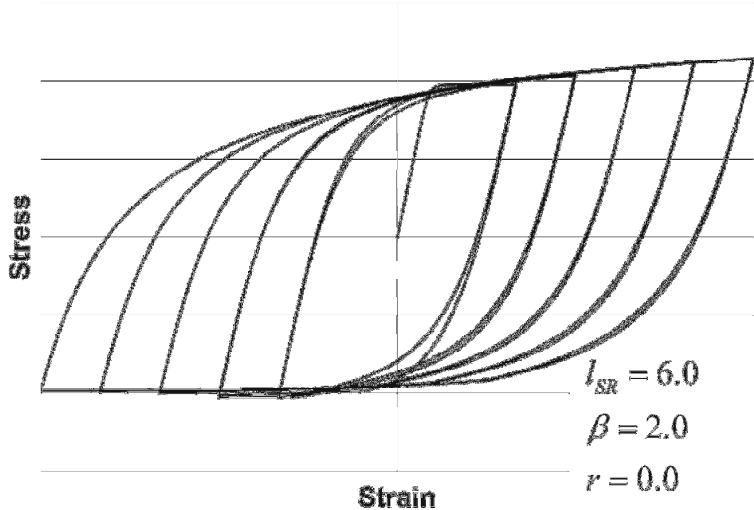
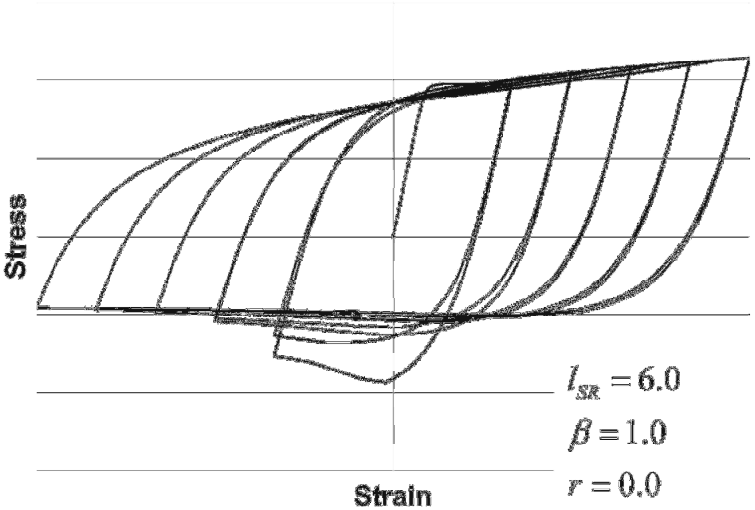
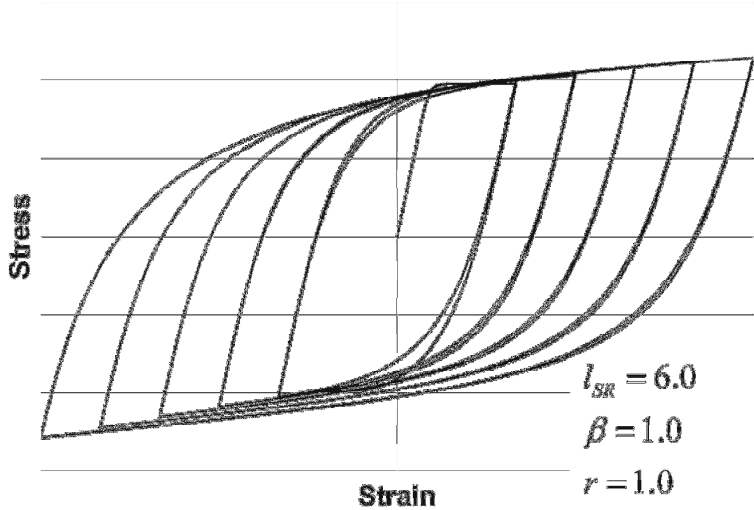
Suggested initial values:

$$\begin{aligned} \beta &= 1.0 \\ r &= 0.4 \\ \gamma &= 0.5 \end{aligned}$$

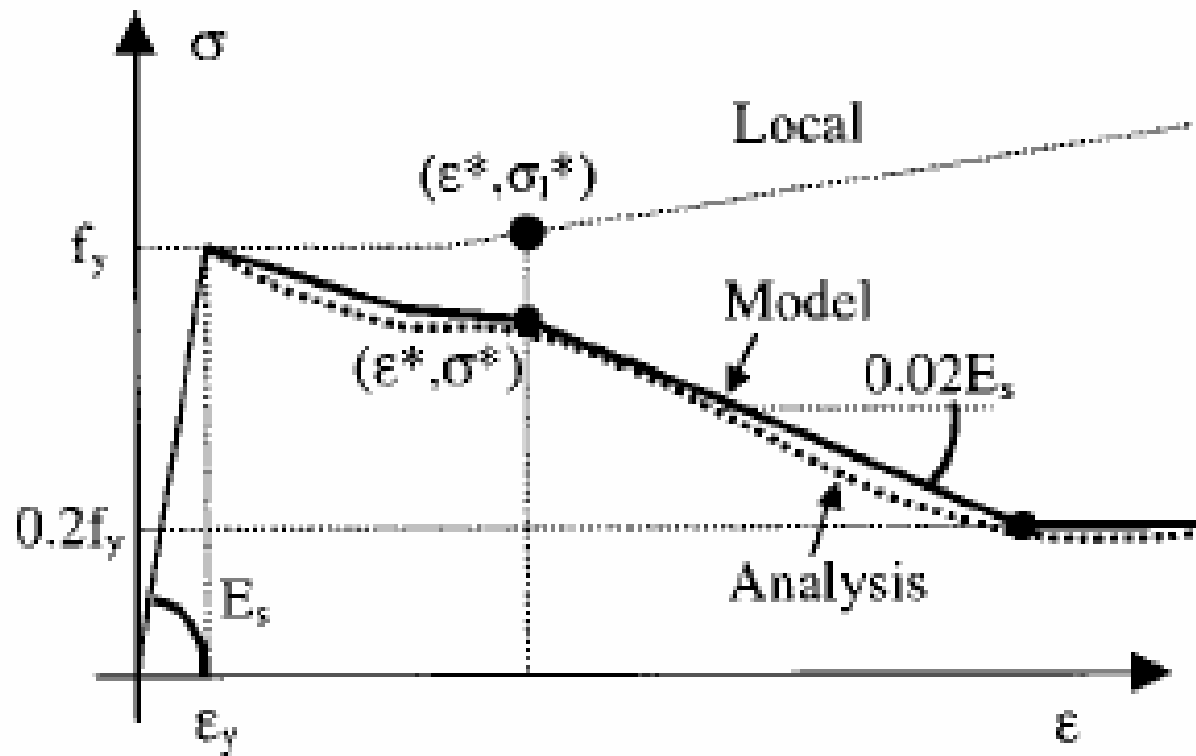


$$\begin{aligned} d_b &= \text{Bar Diameter} \\ L_u &= \text{Unsupported Length} \\ l_{SR} &= \frac{L_u}{d_b} \end{aligned}$$

Gomes Appleton



Dhakal-Maekawa Buckling Behavior



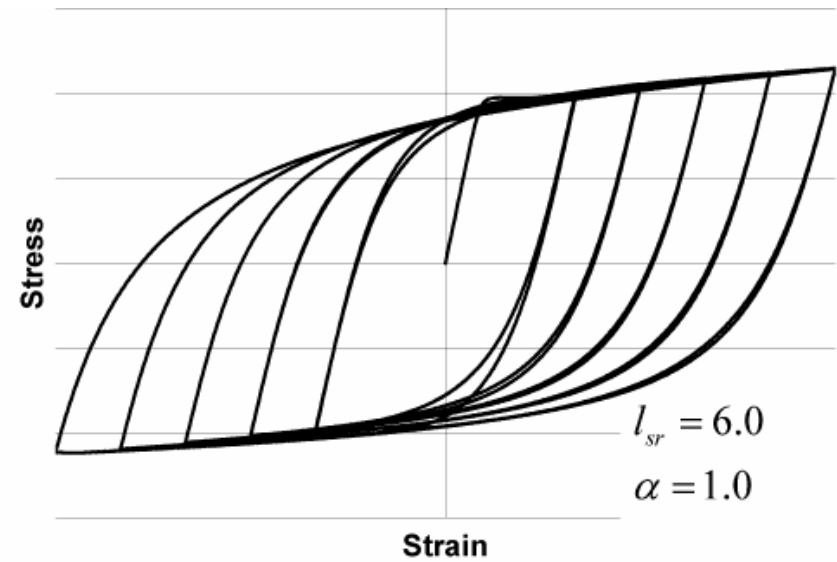
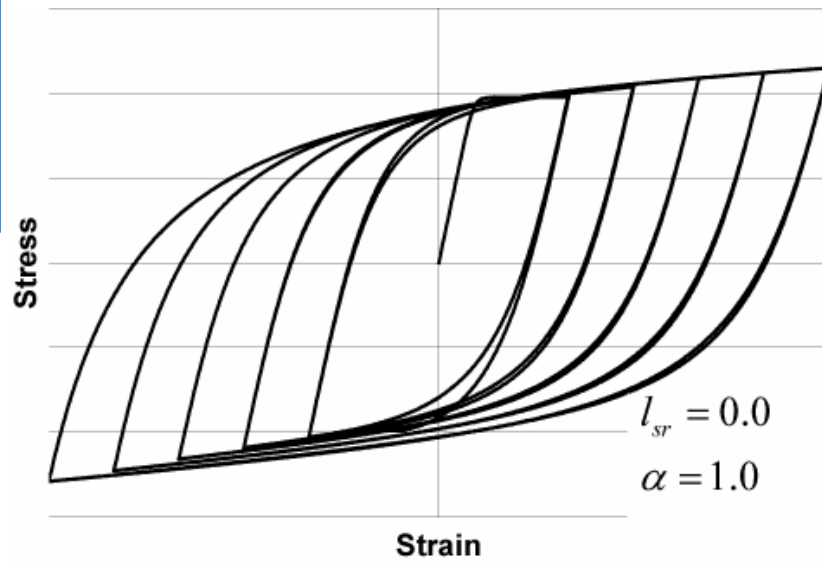
$$\frac{\varepsilon^*}{\varepsilon_y} = 55 - 2.3 \sqrt{\frac{f_y L}{100D}}; \quad \varepsilon^*/\varepsilon_y \geq 7$$

$$\frac{\sigma^*}{\sigma_l^*} = \alpha \left(1.1 - 0.016 \sqrt{\frac{f_y L}{100D}} \right); \quad \sigma^* \geq 0.2f_y$$

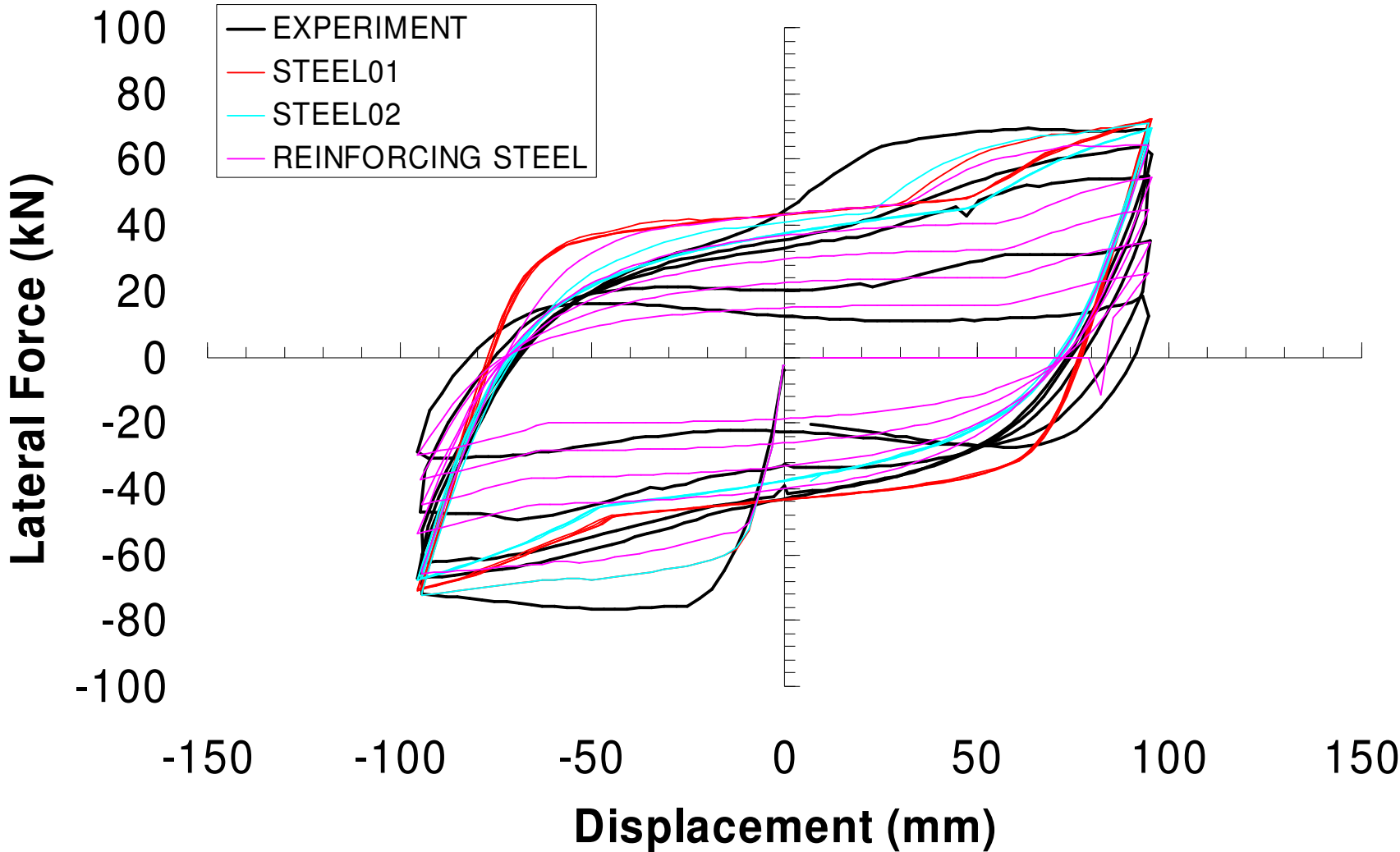
$$\frac{\sigma}{\sigma_l} = 1 - \left(1 - \frac{\sigma^*}{\sigma_l^*} \right) \left(\frac{\varepsilon - \varepsilon_y}{\varepsilon^* - \varepsilon_y} \right); \quad \text{for } \varepsilon_y < \varepsilon \leq \varepsilon^*$$

$$\sigma \geq 0.2f_y; \quad \sigma = \sigma^* - 0.02E_s(\varepsilon - \varepsilon^*); \quad \text{for } \varepsilon > \varepsilon^*$$

Dhakal-Maekawa



Column Simulation – Force Based Beam



OpenSees Implementation

```
uniaxialMaterial ReinforcingSteel $matTag $fy $fu $Es $Esh $esh $eult  
< -GABuck $lsr $beta $r $gama >  
< -DMBuck $lsr < $alpha > >  
< -CMFatigue $Cf $alpha $Cd >  
< -MPCurveParams $MP1 $MP2 $MP3 >
```

Modeling Tips

1. Remember that the material is a single base material with many modifiers. Fatigue and buckling is not automatic.
2. You may get softening in tension but not in compression (without buckling)
3. This material is not symmetrical and will amplify convergence problems. Use less integration points in the NL beam column.
4. Member behavior is mesh biased, more integration points will give higher local curvature. This will increase fatigue, buckling etc. We recommend using beam with hinges so you have control over the hinge length.

Modeling Tips

5. Start with the base model, get it to run, then add the other features one at a time
6. If you get a message “trial strain too large...” you are approaching a compressive strain of 1.0 in one or more fibers, the results are invalid, something is wrong.
7. This material is more complicated and uses more logic and memory. It will be slightly slower than Steel02
- 8.

References

- ◆ Chang, G. and Mander, J. (1994). Seismic Energy Based Fatigue Damage Analysis of Bridge Columns: Part I – Evaluation of Seismic Capacity. NCEER Technical Report 94-0006.
- ◆ Gomes, A., and Appleton, J. (1997). “Nonlinear cyclic stress-strain relationship of reinforcing bars including buckling.” Eng. Struct., 19(10), 822–826.
- ◆ Brown, J. and Kunnath, S.K. (2000). Low Cycle Fatigue Behavior of Longitudinal Reinforcement in Reinforced Concrete Bridge Columns. NCEER Technical Report 00-0007.