

set ORIGIN =1

ORIGIN := 1

Note: Appendix D in the following text corresponds to Appendix D in Kadysiewski and Mosalam, PEER Report, 2009

## Set the units

kip := 1

in := 1

sec := 1

m :=  $\left(\frac{1}{0.0254}\right) \cdot \text{in}$

kN := 0.224808943 · kip

N :=  $\left(\frac{1}{1000}\right) \cdot \text{kN}$

cm := 0.01 · m

mm := 0.001 · m

foot := 0.3048 · m

MPa :=  $1000 \cdot \frac{\text{kN}}{\text{m}^2}$

psi :=  $\frac{\text{MPa}}{145.037738}$

lbf := psi · in · in

ksi := 1000 · psi

g :=  $9.81 \cdot \frac{\text{m}}{\text{sec}^2}$

## Calculation of Infill Properties

fme := 1.00 · ksi Masonry expected compressive strength

tin := 6.00 · in Thickness of masonry infill wall

hin := 140.0 · in Height of masonry infill wall

$L_{inf} := 120.0\text{-in}$  Length of masonry infill wall

$h_{col} := 140\text{-in}$  Floor-to-floor height

$L_{col} := 120.0\text{-in}$  Centerline distance between columns

$E_m := 550\text{-fme}$  FEMA 356 formula for masonry elastic modulus  
(expected and lower bound, unit: psi, Tables 7-1 and 7-2  
You can input  $E_m$  separately as well

$E_m := 500\text{-ksi}$

$E_{fc} := 3122\text{-ksi}$  Expected elastic modulus of frame concrete

$I_g := 8856\text{-in}^4$  Gross moment of inertia of the concrete columns

$I_{col} := 0.5 \cdot I_g$  Effective cracked moment of inertia of the concrete columns

$$I_{col} = 4.428 \times 10^3$$

$r_{inf} := \sqrt{h_{inf}^2 + L_{inf}^2}$  Diagonal length of the infill

$$r_{inf} = 184.391$$

$\theta_{inf} := \text{atan}\left(\frac{h_{inf}}{L_{inf}}\right)$  Angle of the diagonal for the infill

$$\theta_{inf} = 0.862$$

$L_{diag} := \sqrt{h_{col}^2 + L_{col}^2}$  Diagonal length between column centerlines and floor centerlines

$$L_{diag} = 184.391$$

$\theta_{diag} := \text{atan}\left(\frac{h_{col}}{L_{col}}\right)$  Angle of the diagonal between beam-column workpoints

$$\theta_{diag} = 0.862$$

$\Gamma_w := 1.2732$  PEER 2008/102, D-5

### Calculate the axial stiffness of the infill strut:

Calculate the width of the compression strut which represents the infill, based on the method given in FEMA 356, Section 7.5.2

$$\lambda_1 := \left( \frac{E_m \cdot t_{inf} \cdot \sin(2 \cdot \theta_{inf})}{4 \cdot E_{fe} \cdot I_{col} \cdot h_{inf}} \right)^{\frac{1}{4}}$$

$$\lambda_1 = 0.025$$

$$a := 0.175 \cdot (\lambda_1 \cdot h_{col})^{-0.4} \cdot r_{inf}$$

$$a = 19.589$$

$$k_{inf} := \frac{a \cdot t_{inf} \cdot E_m}{r_{inf}}$$

$$k_{inf} = 318.708$$

Calculate the required area of the equivalent element, which will span between workpoints and will have an elastic modulus equal to  $E_m$

$$A_{elem} := \frac{k_{inf} \cdot L_{diag}}{E_m}$$

$$A_{elem} = 117.534$$

### Calculate the axial strength of the infill strut (Based on FEMA356, Section 7.5.2.2)

$P_{ce} := 41.4 \cdot \text{kip}$  Expected gravity compressive force applied to infill panel

$v_{te} := 900 \cdot \text{psi}$  Average bed joint strength

$A_n := t_{inf} \cdot L_{inf}$  Net bedded area of the infill

$$A_n = 720$$

$v_{me} := \frac{0.75 \cdot \left( v_{te} + \frac{P_{ce}}{A_n} \right)}{1.5}$  Expected masonry shear strength

$$v_{me} = 0.479$$

$f_{vie} := 0.05 \cdot \text{ksi}$

$v_{shear} := \min(v_{me}, f_{vie})$

$$v_{shear} = 0.05$$

$Q_{ce} := v_{shear} \cdot A_n$  Expected horizontal shear capacity of infill

$$Q_{ce} = 36$$

$$P_{n0} := \frac{Q_{ce}}{\cos(\theta_{diag})} \quad \text{Axial capacity of the equivalent compression strut}$$

$$P_{n0} = 55.317$$

### Calculate the in-plane displacement properties of the infill strut:

Calculate the "yield point", i.e., the axial deformation in the equivalent strut at the point where the initial tangent stiffness line intersects the element capacity:

$$\delta_{Ay0} := \frac{P_{n0}}{k_{inf}}$$

assumes no OOP load

$$\delta_{Ay0} = 0.174$$

Calculate the IP horizontal deflection of the panel at the yield point:

$$u_{Hy0} := \frac{\delta_{Ay0}}{\cos(\theta_{diag})}$$

$$u_{Hy0} = 0.267 \quad \text{Note : assumes that the vertical deflections of the endpoints are zero}$$

Calculate the lateral deflection of the panel at the collapse prevention (CP) limit state:  
Based on FEMA 356, Section 7.5.3.2.4, including Table 7-9:

1) Estimate  $0.7 < \beta < 1.3$ , where  $\beta$  is defined as  $V_{fre}/V_{ine}$ , the ratio of frame to infill expected strengths

$$2) \frac{L_{inf}}{h_{inf}} = 0.857$$

3) 

0.5	1%
1	0.8%
2	0.6%

 Interpolated in Table 7-9. It is assumed that the CP limit state is reached when the element drift reaches point "d" as shown in Figure 7.1 of FEMA356

$$d := \text{linterp}\left(\text{table79}^{(1)}, \text{table79}^{(2)}, \frac{L_{inf}}{h_{inf}}\right)$$

$$d = 0.00857$$

$$u_{Hcp0} := d \cdot h_{inf} \quad \text{Displacement of the panel at the limit state}$$

$$u_{Hcp0} = 1.2$$

$$\mu_{H0} := \frac{u_{Hcp0}}{u_{Hy0}} \quad \text{Implied ductility at the collapse prevention level}$$

$$\mu_{H0} = 4.499$$

**Calculate the Out-of-Plane (OOP) parameters of the infill:**

$$\gamma_{\text{inf}} := 15 \cdot \frac{\text{kN}}{\text{m}^3}$$

Weight density of the infill bricks (assumed).

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Calculate the OOP frequency of the infill, assuming that it spans vertically, with simply-supported ends:

$$I_{\text{inf\_g}} := \frac{L_{\text{inf}} \cdot t_{\text{inf}}^3}{12} \quad \text{Moment of inertia of the uncracked infill (gross moment)}$$

$$I_{\text{inf}} := \frac{1}{2} \cdot I_{\text{inf\_g}} \quad \text{Estimated moment of inertia of the cracked infill}$$

$$I_{\text{inf}} = 1.08 \times 10^3$$

$$w_{\text{inf}} := L_{\text{inf}} \cdot t_{\text{inf}} \cdot \gamma_{\text{inf}} \quad \text{Weight per unit of length (measured vertically) of the infill.}$$

$$w_{\text{inf}} = 0.04$$

$$f_{\text{ss}} := \frac{\pi}{2 \cdot h_{\text{inf}}^2} \cdot \sqrt{\frac{E_{\text{m}} \cdot I_{\text{inf}} \cdot g}{w_{\text{inf}}}} \quad \text{First natural frequency of the infill, spanning in the vertical direction, with top and bottom ends simply supported. (Blevins, 1979, Table 8-1).}$$

$$f_{\text{ss}} = 5.802 \quad \text{per} := \frac{1}{f_{\text{ss}}}$$

$$\text{per} = 0.1723$$

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Calculate the OOP effective weight:

The OOP effective weight is based on the modal effective mass of the vertically spanning, simply supported (assumed) infill wall. For simple-simple conditions, the modal effective weight is equal to 81% of the total infill weight. See Appendix D for a derivation of this value.

$$W_{\text{inf}} := \gamma_{\text{inf}} \cdot t_{\text{inf}} \cdot h_{\text{inf}} \cdot L_{\text{inf}} \quad \text{Total weight of the infill.}$$

$$W_{\text{inf}} = 5.57$$

$$\text{MEW} := 0.81 \cdot W_{\text{inf}} \quad \text{Modal effective weight, assuming that the wall spans vertically, is simply supported top and bottom.}$$

$$\text{MEW} = 4.512 \quad \text{(First mode). See Appendix D. This value divide by } g \text{ is the OOP mass value that should be assigned to the center node}$$

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Calculate the equivalent OOP spring which will provide the identical frequency.

$$k_{eq\_N} := (2 \cdot \pi \cdot f_{ss})^2 \cdot \frac{MEW}{g}$$

$$k_{eq\_N} = 15.527$$

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Calculate the moment of inertia of the equivalent beam element, such that it will provide the correct value of  $k_{eq\_N}$ :

$$I_{eq} := \frac{k_{eq\_N} (L_{diag})^3}{48 \cdot E_m}$$

$$I_{eq} = 4056.03129 \qquad I_{elem} := I_{eq} \qquad \frac{a \cdot t_{inf}^3}{12} = 352.601$$

Using Equation D.28 from Appendix D:

$$1.644 \cdot \left( \frac{L_{diag}}{h_{inf}} \right)^3 \cdot t_{inf} = 4056.6 \qquad \text{(Same results)}$$

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Calculate the OOP Capacity of the infill:

The OOP capacity is based on FEMA 356, Section 7.5.3.2.

$$\frac{h_{inf}}{t_{inf}} = 23.333$$

Since this value is outside the range used in FEMA 356, Table 7-11, for determining  $\lambda$ , perform an extrapolation:

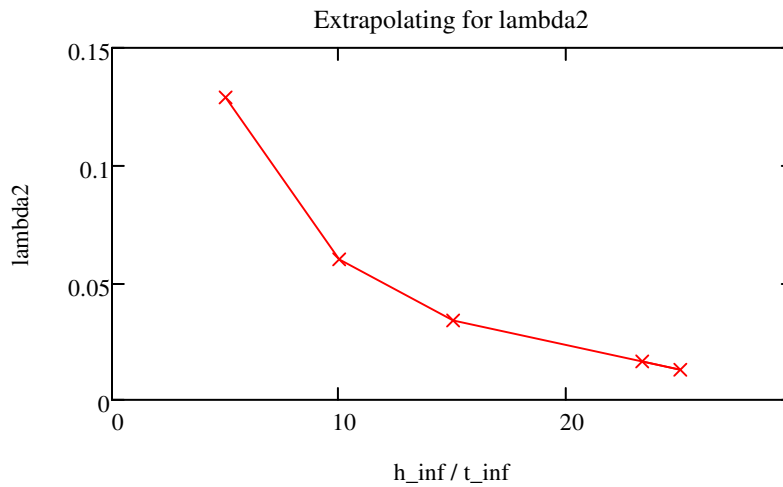
Array of values from Table 7-11:

$$FEMA\_Array := \begin{pmatrix} 5 & 0.129 \\ 10 & 0.060 \\ 15 & 0.034 \\ 25 & 0.013 \end{pmatrix}$$

$$\lambda_2 := \text{linterp}\left(\text{FEMA\_Array}^{(1)}, \text{FEMA\_Array}^{(2)}, \frac{h_{inf}}{t_{inf}}\right) \quad \lambda_2 = 0.01650$$

Check by graphing the values:

$$\text{Array} := \begin{pmatrix} 5 & 0.129 \\ 10 & 0.060 \\ 15 & 0.034 \\ 25 & 0.013 \\ \frac{h_{inf}}{t_{inf}} & \lambda_2 \end{pmatrix} \quad k := 1.. \text{rows}(\text{Array})$$



$$q_{in} := \frac{0.7 \cdot f_{me} \cdot \lambda_2}{\frac{h_{inf}}{t_{inf}}}$$

Note: the expected, rather than the lower bound value, of masonry compressive strength is used here, since the expected OOP strength will be used in later calculations.

$$q_{in} = 0.000495$$

$$q_{in} = 0.495 \text{ psi}$$

$$q_{in} \cdot h_{inf} \cdot L_{inf} = 8.316$$

Total OOP force on the wall at capacity.



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Calculate the moment in the infill wall at the time that it reaches its capacity:

$$M_y := \frac{q_{in} \cdot L_{inf} \cdot h_{inf}^2}{8} \quad \text{Assumes simple support at the top and bottom.}$$

$$M_y = 145.53$$

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Calculate the required yield moment for the equivalent element, such that the same base motion will bring it and the original wall to incipient yield:

$$M_{eq\_y} := 1.570 \cdot \frac{L_{diag}}{h_{inf}} \cdot M_y \quad \text{Note: for derivation of this equation, see Appendix D.}$$

$$M_{eq\_y} = 300.929 \cdot \text{in} \cdot \text{kip}$$

$$M_{n0} := M_{eq\_y} \quad \text{Defines the OOP "yield" moment for the equivalent member when the IP axial force is zero.}$$

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Determine the OOP point force, applied at the midspan of the equivalent element, to cause yielding:

$$F_{Ny0} := \frac{4 \cdot M_{eq\_y}}{L_{diag}}$$

$$F_{Ny0} = 6.528$$

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Calculate the displacement of the equivalent element at first yield and at the collapse prevention limit state, assuming no IP axial force:

$$u_{Ny0} := \frac{F_{Ny0}}{k_{eq\_N}} \quad \text{OOP "yield" displacement, assuming no IP axial force.}$$

$$u_{Ny0} = 0.42 \cdot \text{in}$$

The displacement at collapse prevention limit state:

FEMA 356, Section 7.5.3.3 gives a maximum OOP deflection based on an OOP story drift ratio of 5%.

$$u_{Ncp0} := 0.05 \cdot h_{inf}$$

$$u_{Ncp0} = 7$$

This value seems too high, since it's larger than the thickness of the infill itself. Instead, define the CP displacement as equal to one half the thickness of the infill.

$$u_{Ncp0} := \min\left(0.05 \cdot h_{inf}, \frac{t_{inf}}{2}\right)$$

$$u_{Ncp0} = 3 \cdot in$$

The implied ductility ratio is:

$$\mu_{Ncp0} := \frac{u_{Ncp0}}{u_{Ny0}}$$

$$\mu_{Ncp0} = 7.136$$

This ductility still seems too high. Based on judgment, use a (conservative) ductility of 5:

$$\mu_{Ncp0} := 5$$

$$u_{Ncp0} := u_{Ny0} \cdot \mu_{Ncp0}$$

$$u_{Ncp0} = 2.102 \cdot in$$

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Calculating the axial force - moment interaction curve for specific values of  $P_{n0}$  and  $M_{n0}$ :

Using the exponent relationship:

$$f_{P_n}(M_n, P_{n0}, M_{n0}) := P_{n0} \cdot \left[ 1 - \left( \frac{M_n}{M_{n0}} \right)^{\frac{3}{2}} \right]^{\frac{2}{3}}$$

This is the target P-M relationship for the equivalent member, located on the diagonal between structural workpoints.

$P_{n0} = 55.317 \cdot \text{kip}$  Axial capacity of the member under pure compression (calculated above).

$M_{n0} = 300.929 \cdot \text{in} \cdot \text{kip}$  Moment capacity of the member under pure bending (calculated above).

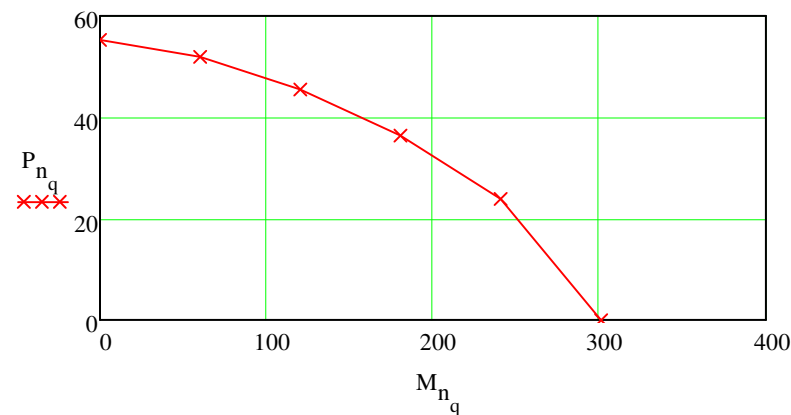
$N_{\text{interaction}} := 6$  Number of points on the interaction curve to be used for calculating fiber properties (should be an even number).

$$N_{\text{fiber}} := 2 \cdot (N_{\text{interaction}} - 1) \quad N_{\text{fiber}} = 10$$

$$M_n := \begin{cases} \text{for } q \in 1..N_{\text{interaction}} \\ M_{n_q} \leftarrow (q - 1) \cdot \frac{M_{n0}}{N_{\text{interaction}} - 1} \\ M_n \end{cases} \quad P_n := \begin{cases} \text{for } q \in 1..N_{\text{interaction}} \\ P_{n_q} \leftarrow f_{P_n}(M_{n_q}, P_{n0}, M_{n0}) \\ P_n \end{cases}$$

$$M_n^T = (0 \ 60.186 \ 120.371 \ 180.557 \ 240.743 \ 300.929) \quad P_n^T = (55.317 \ 51.968 \ 45.542 \ 36.467 \ 23.926 \ 0)$$

$$q := 1..N_{\text{interaction}}$$



Calculate the required strength and location of the various fibers:

$$F_y := \begin{cases} \text{for } p \in 1..N_{\text{interaction}} - 1 \\ F_{y_p} \leftarrow \frac{P_{n_p} - P_{n_{p+1}}}{2} \\ \text{for } p \in N_{\text{interaction}}..2 \cdot (N_{\text{interaction}} - 1) \\ F_{y_p} \leftarrow F_{y_2} \cdot N_{\text{interaction}}^{-1-p} \\ F_y \end{cases}$$

	1
1	1.675
2	3.213
3	4.538
4	6.27
5	11.963
6	11.963
7	6.27
8	4.538
9	3.213
10	1.675

$$\sum_{p=1}^{N_{\text{fiber}}} F_{y_p} = 55.317 \cdot \text{kip}$$

$$z := \begin{cases} \text{for } p \in 1..N_{\text{interaction}} - 1 \\ z_p \leftarrow \frac{M_{n_{p+1}} - M_{n_p}}{2 \cdot F_{y_p}} \\ \text{for } p \in N_{\text{interaction}}..2 \cdot (N_{\text{interaction}} - 1) \\ z_p \leftarrow -(z_2 \cdot N_{\text{interaction}}^{-1-p}) \\ z \end{cases}$$

	1
1	17.967
2	9.367
3	6.631
4	4.799
5	2.515
6	-2.515
7	-4.799
8	-6.631
9	-9.367
10	-17.967

$$\text{abs}(x) := \text{if}(x \geq 0.0, x, -x)$$

Absolute function (since the MathCad absolute function has some bugs).

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Solve block for the determining the values of the parameters  $\gamma$  and  $\eta$ :

Estimate the values of the parameters:  $\gamma := 30$        $\eta := -2$

Given

$$\sum_{p=1}^{N_{\text{fiber}}} \left[ \gamma \cdot (\text{abs}(z_p))^\eta \right] = A_{\text{elem}} \qquad \sum_{p=1}^{N_{\text{fiber}}} \left[ \left[ \gamma \cdot (\text{abs}(z_p))^\eta \right] \cdot (z_p)^2 \right] = I_{\text{elem}}$$

Result := Find( $\gamma, \eta$ )

$$\underline{\gamma} := \text{Result}_1 \cdot \text{in}^2 \qquad \gamma = 102.789 \cdot \text{in}^2$$

$$\underline{\eta} := \text{Result}_2 \qquad \eta = -1.329$$

$$\underline{A} := \begin{cases} \text{for } p \in 1..N_{\text{fiber}} \\ A_p \leftarrow \gamma \cdot (\text{abs}(z_p))^\eta \\ A \end{cases}$$

$$A^T =$$

	1	2	3	4	5	6	7	8
1	2.214	5.261	8.324	12.791	30.177	30.177	12.791	...

Check the results above:

$$\sum_{p=1}^{N_{\text{fiber}}} A_p = 117.534$$

$$\sum_{p=1}^{N_{\text{fiber}}} \left[ A_p \cdot (z_p)^2 \right] = 4056.0313$$

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Determine the stress at yield:

$$\sigma_y := \begin{cases} \text{for } p \in 1..N_{\text{fiber}} \\ \sigma_p \leftarrow \frac{F_{y_p}}{A_p} \\ \sigma \end{cases}$$

$$\sigma_y^T =$$

	1	2	3	4	5	6	7	8	9	10
1	0.756	0.611	0.545	0.49	0.396	0.396	0.49	0.545	0.611	0.756

Calculate the strain at first yield:

$$\varepsilon_y := \begin{cases} \text{for } p \in 1..N_{\text{fiber}} \\ \varepsilon_{y_p} \leftarrow \frac{\sigma_{y_p}}{E_m} \\ \varepsilon_y \end{cases}$$

$$\varepsilon_y^T =$$

	1	2	3	4	5
1	$1.513 \cdot 10^{-3}$	$1.221 \cdot 10^{-3}$	$1.09 \cdot 10^{-3}$	$9.804 \cdot 10^{-4}$	...

$$\text{Ratio} := \begin{cases} \text{for } p \in 1..N_{\text{fiber}} \\ \text{Ratio}_p \leftarrow \frac{\varepsilon_{y_p}}{z_p} \\ \text{Ratio} \end{cases}$$

$$\text{Ratio}^T =$$

	1	2	3	4	5	6
1	$8.42 \cdot 10^{-5}$	$1.304 \cdot 10^{-4}$	$1.644 \cdot 10^{-4}$	$2.043 \cdot 10^{-4}$	$3.152 \cdot 10^{-4}$	...

## Summary of Fiber Properties:

Elastic Modulus:  $E_m = 500$

Fiber yield strength:

	1
1	1.675
2	3.213
3	4.538
4	6.27
$F_y =$ 5	11.963
6	11.963
7	6.27
8	4.538
9	3.213
10	1.675

Fiber Area:

	1
1	2.214145
2	5.260806
3	8.324044
4	12.79111
$A =$ 5	30.176721
6	30.176721
7	12.79111
8	8.324044
9	5.260806
10	2.214145

Fiber location (distance from CL):

	1
1	17.967402
2	9.367025
3	6.63148
4	4.799369
$z =$ 5	2.515478
6	-2.515478
7	-4.799369
8	-6.63148
9	-9.367025
10	-17.967402

Fiber yield stress:

	1
1	0.756
2	0.611
3	0.545
4	0.49
$\sigma_y =$ 5	0.396
6	0.396
7	0.49
8	0.545
9	0.611
10	0.756

Fiber yield strain:

	1
1	$1.513 \cdot 10^{-3}$
2	$1.221 \cdot 10^{-3}$
3	$1.09 \cdot 10^{-3}$
4	$9.804 \cdot 10^{-4}$
$\epsilon_y =$ 5	$7.929 \cdot 10^{-4}$
6	$7.929 \cdot 10^{-4}$
7	$9.804 \cdot 10^{-4}$
8	$1.09 \cdot 10^{-3}$
9	$1.221 \cdot 10^{-3}$
10	$1.513 \cdot 10^{-3}$

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Verify that the given parameters will produce the desired section properties:

$$A_{\text{calc}} := \sum_{p=1}^{N_{\text{fiber}}} A_p \quad A_{\text{calc}} = 117.534 \cdot \text{in}^2 \quad \frac{A_{\text{calc}}}{A_{\text{elem}}} = 1.000$$

$$I_{\text{calc}} := \sum_{p=1}^{N_{\text{fiber}}} \left[ A_p \cdot (z_p)^2 \right] \quad I_{\text{calc}} = 4056.0313 \cdot \text{in}^4 \quad \frac{I_{\text{calc}}}{I_{\text{elem}}} = 1.000$$

$$P_{0\_calc} := \sum_{p=1}^{N_{\text{fiber}}} \left( A_p \cdot \sigma_{y_p} \right) \quad P_{0\_calc} = 55.317 \cdot \text{kip} \quad \frac{P_{0\_calc}}{P_{n0}} = 1.000$$

$$M_{0\_calc} := \sum_{p=1}^{N_{\text{fiber}}} \left( \sigma_{y_p} \cdot A_p \cdot \text{abs}(z_p) \right) \quad M_{0\_calc} = 300.929 \cdot \text{in} \cdot \text{kip} \quad \frac{M_{0\_calc}}{M_{n0}} = 1.000$$

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Calculating the IP disp - OOP disp curve for specific values of OOP disp

Using the exponent relationship:

$$f\_OOP(OOP, IP0, OOP0) := IP0 \cdot \left[ 1 - \left( \frac{OOP}{OOP0} \right)^{\frac{3}{2}} \right]^{\frac{2}{3}}$$

This is the target P-M relationship for the equivalent member, located on the diagonal between structural workpoints.

$Pn0 = 55.317 \cdot \text{kip}$  Axial capacity of the member under pure compression (calculated above).

$Mn0 = 300.929 \cdot \text{in} \cdot \text{kip}$  Moment capacity of the member under pure bending (calculated above).

$N_{\text{interaction}} := 10$  Number of points on the interaction curve to be used for calculating fiber properties (should be an even number).

$$OOPv := \begin{cases} \text{for } q \in 1..N_{\text{interaction}} \\ OOPv_q \leftarrow (q - 1) \cdot \frac{u_{Ncp0}}{N_{\text{interaction}} - 1} \\ OOPv \end{cases}$$

$$IIPv := \begin{cases} \text{for } q \in 1..N_{\text{interaction}} \\ IIPv_q \leftarrow f\_OOP(OOPv_q, u_{Hcp0}, u_{Ncp0}) \\ IIPv \end{cases}$$

$$OOPv^T =$$

	1	2	3	4	5	6	7	8	9	10
1	0	0.234	0.467	0.701	0.934	1.168	1.401	1.635	1.869	2.102

$$IIPv^T =$$

	1	2	3	4	5	6	7	8	9	10
1	1.2	1.17	1.115	1.041	0.949	0.84	0.711	0.554	0.357	0

$$q := 1..N_{\text{interaction}}$$

